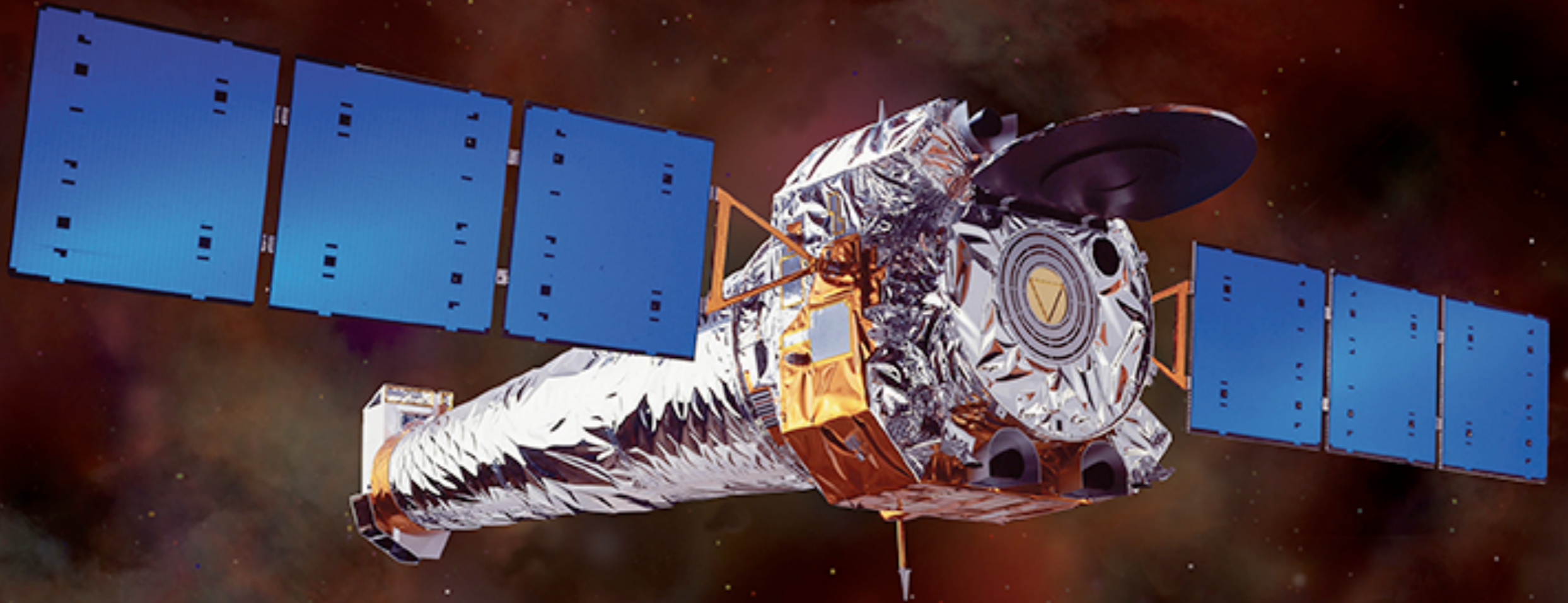


The Chandra X-ray Observatory

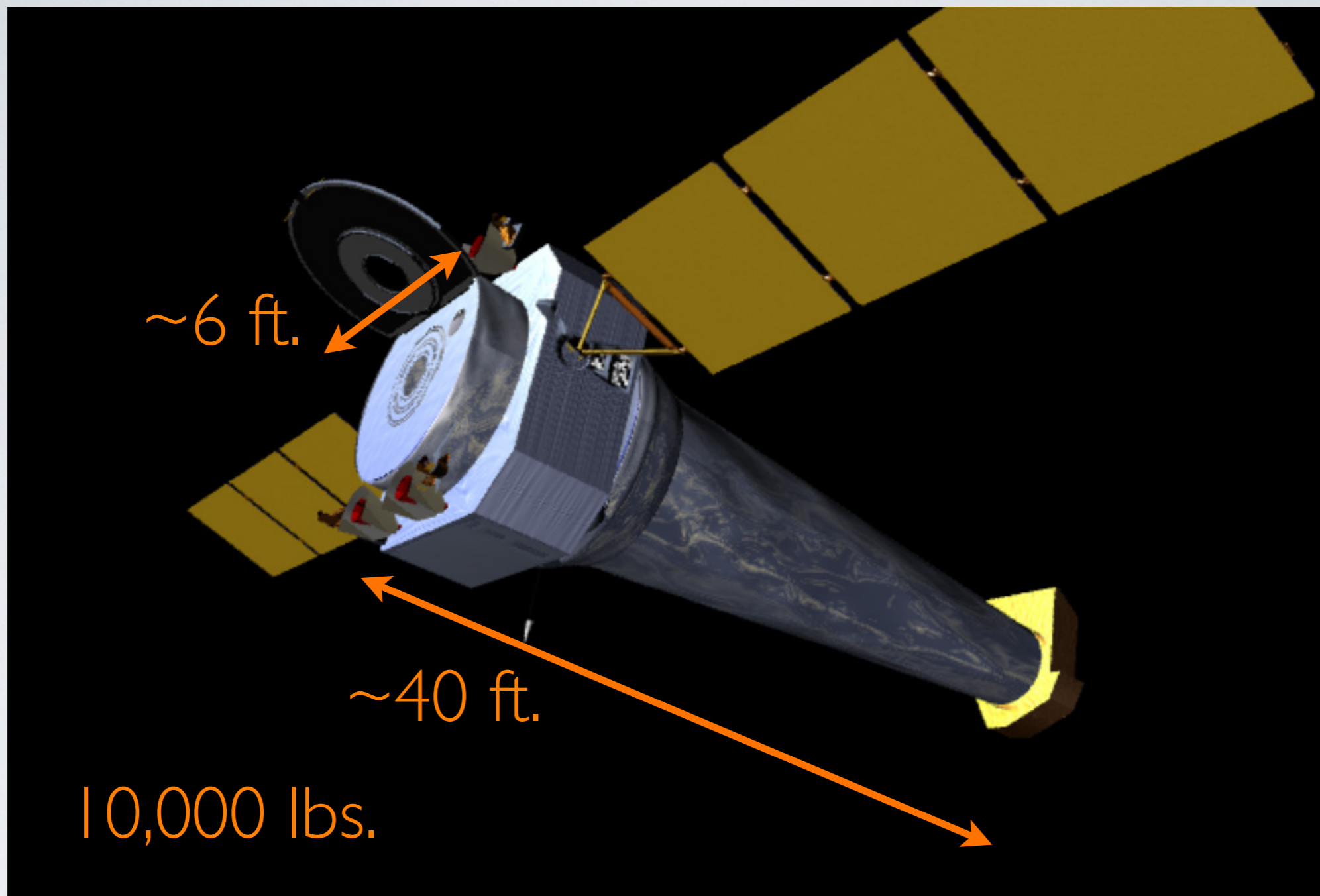


Instruments and Operations

Dr. Michael Nowak
Chandra X-ray Science Center

UNIQUE CHANDRA ATTRIBUTES:

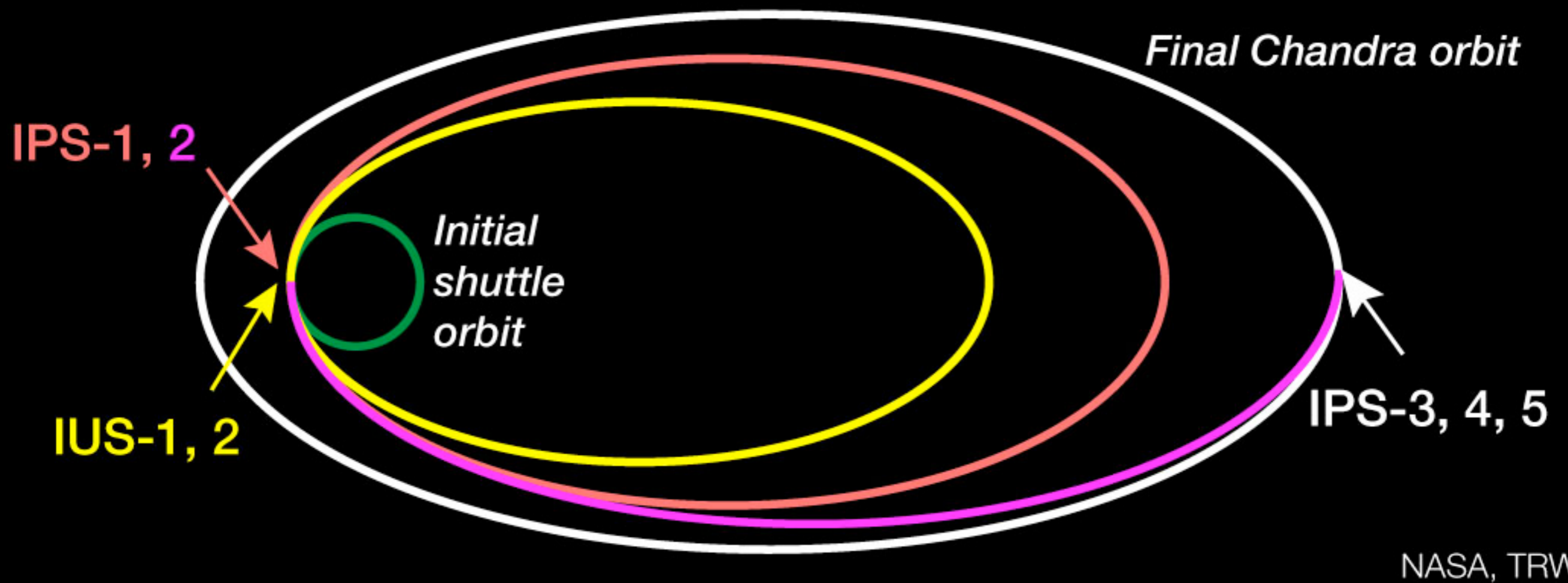
- Best Spatial Resolution of Any X-ray Satellite
 - $\sim 1''$ (Hubble $\sim 0.1''$, next best X-ray satellite, XMM-Newton $\sim 10''$, ROSAT $\sim 5''$)
 - This is coupled with good energy range & resolution – we view 300 eV – 9 keV, $E/\Delta E \sim 5 - 40$
- Best Energy Resolution (Gratings) of Any X-ray Satellite
 - $E/\Delta E \sim 1400 - 200$ (Radio & Hubble $\sim 20,000$, next best X-ray satellite, XMM-Newton $\sim 500 - 40$)
- Largest Dynamic Flux Range of Any Satellite Ever Flown:
11 Orders of Magnitude; $10^{-18} - 10^{-7} \text{ erg cm}^{-2} \text{ s}^{-1}$



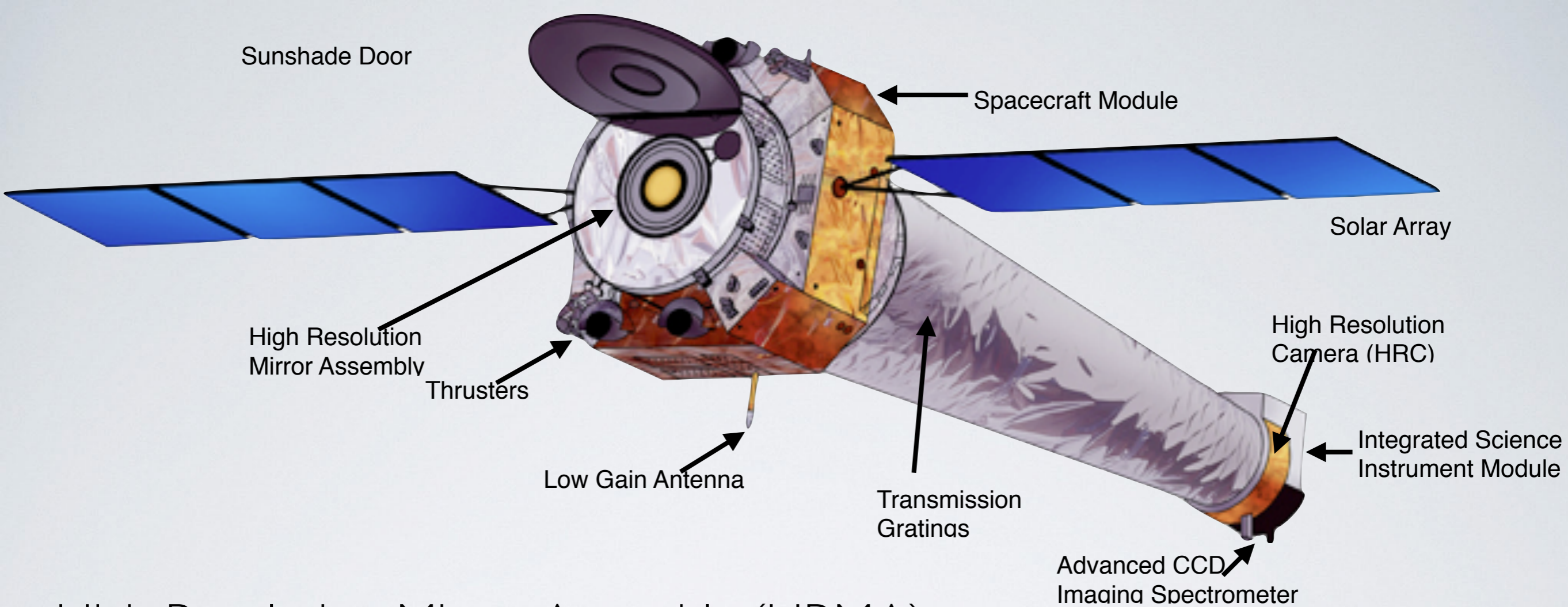
LARGEST SATELLITE LAUNCHED
BY SPACE SHUTTLE:
JULY 23, 1999 – 2030 (?)



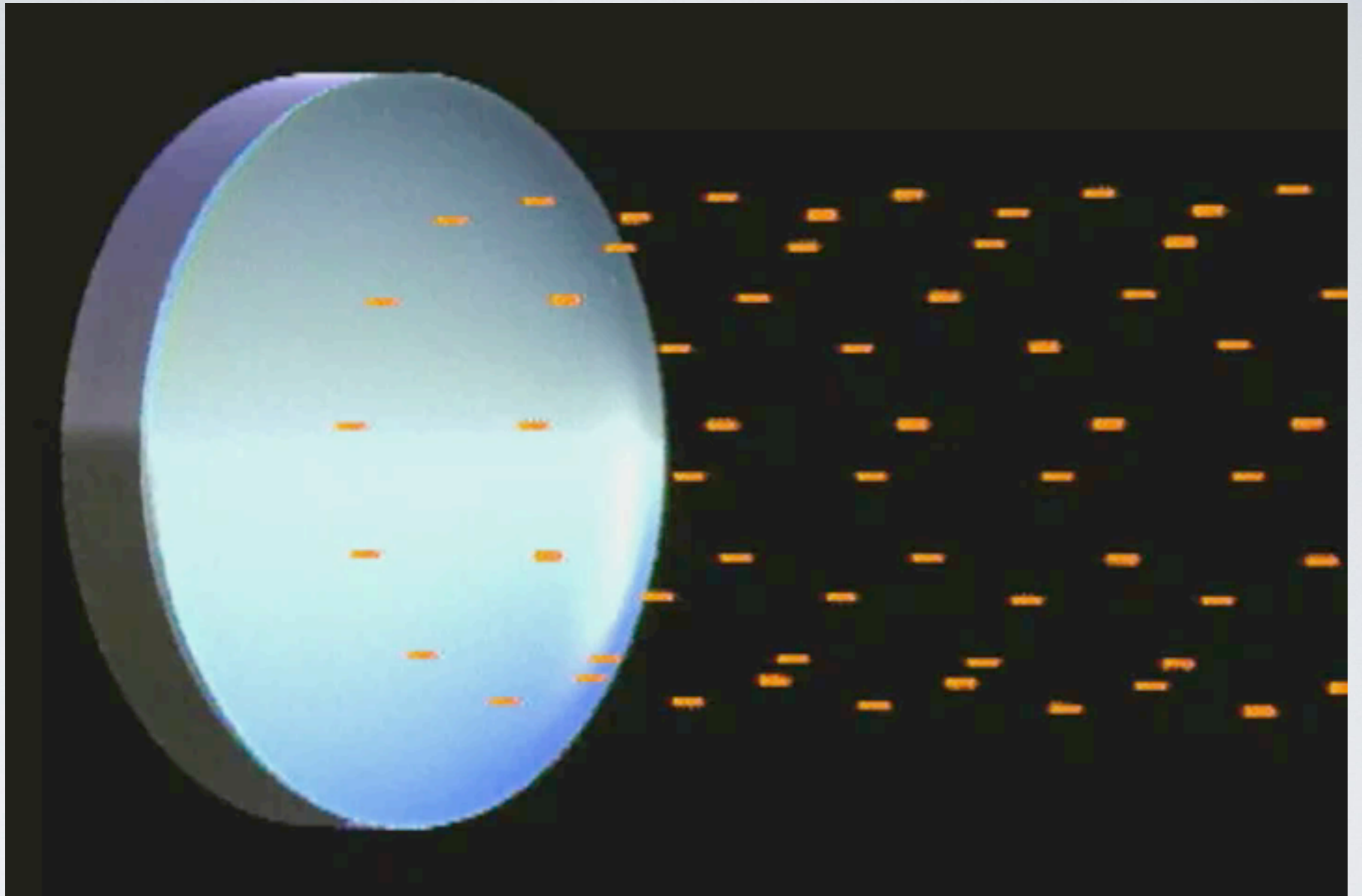
BOOSTED INTO ELLIPTICAL
ORBIT: 229 KSEC=63.5 HOURS



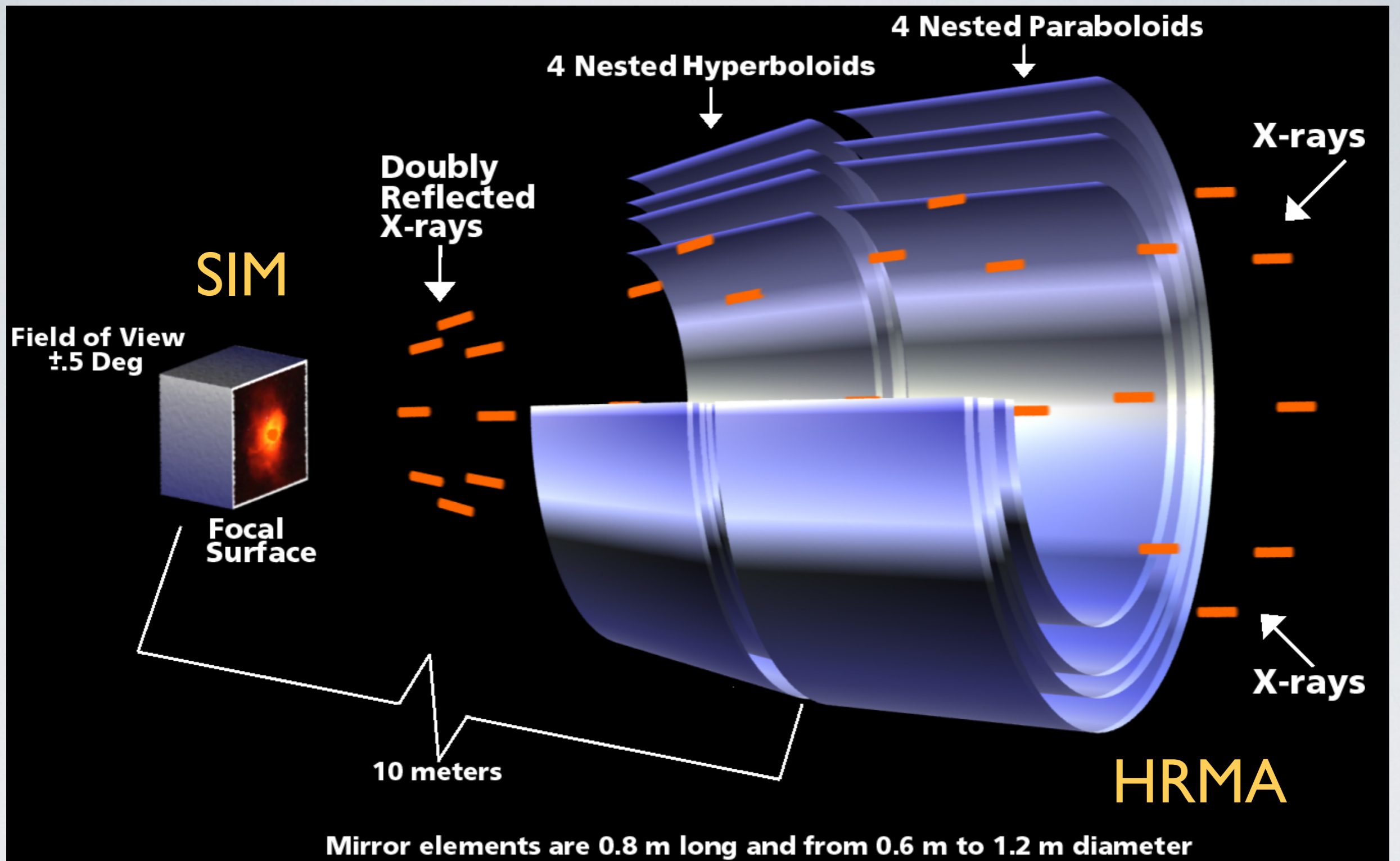
CONTINUOUS OBSERVATIONS AS
LONG AS 170 KSEC
IN PRACTICE, USUALLY SHORTER
(\approx 20 KSEC, EASY TO DO)



- High Resolution Mirror Assembly (HRMA)
- Scientific Instrument Module (SIM)
 - Advanced CCD Imaging Spectrometer (ACIS)
 - High Resolution Camera (HRC)
- Transmission Gratings: HETG and LETG
- ACIS-S, ACIS-I Imaging
- HETG/ACIS-S, LETG/ACIS-S
- HRC-I/-S Imaging & Timing
- LETG/HRC-S

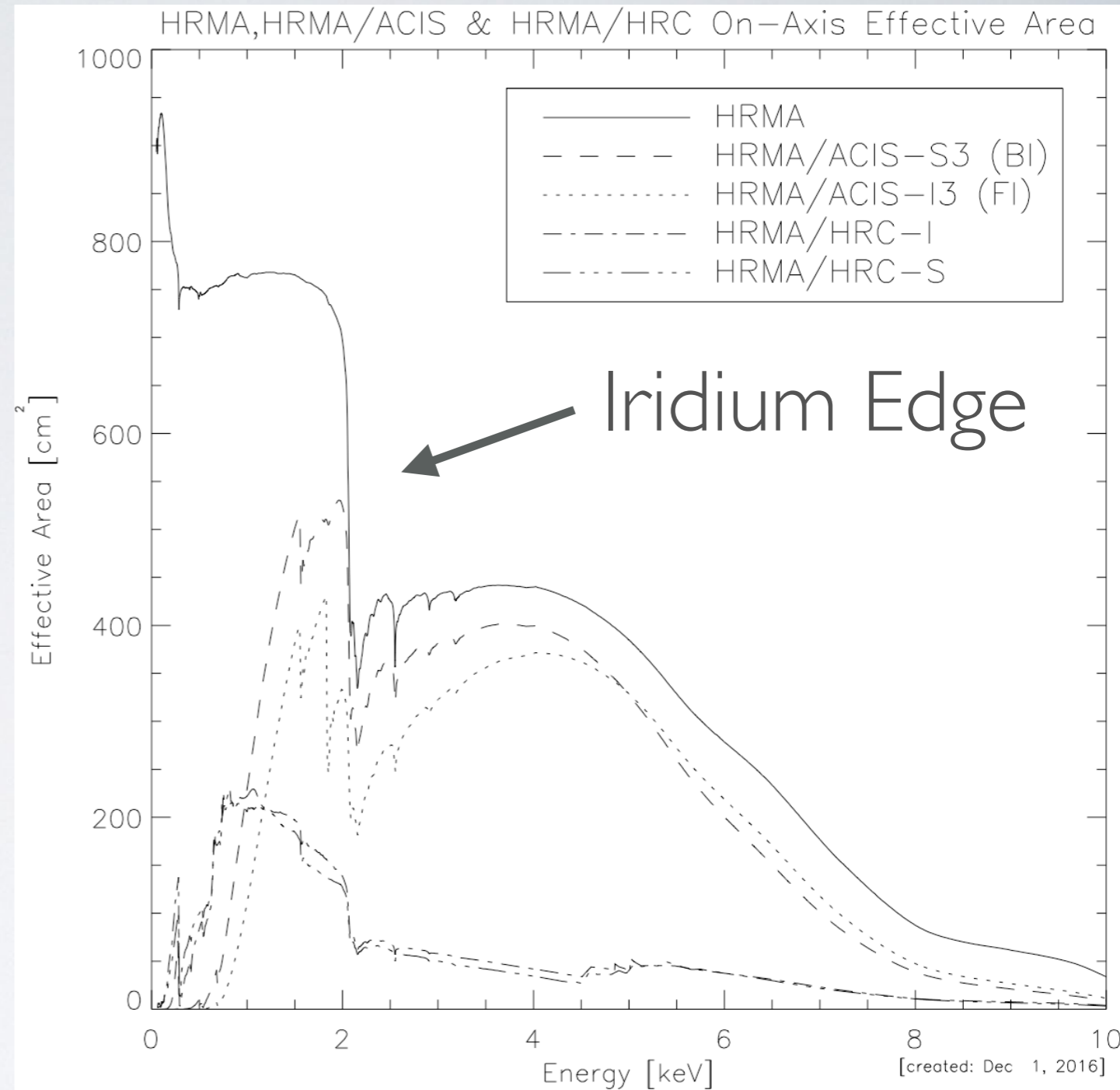


CHANDRA MIRRORS

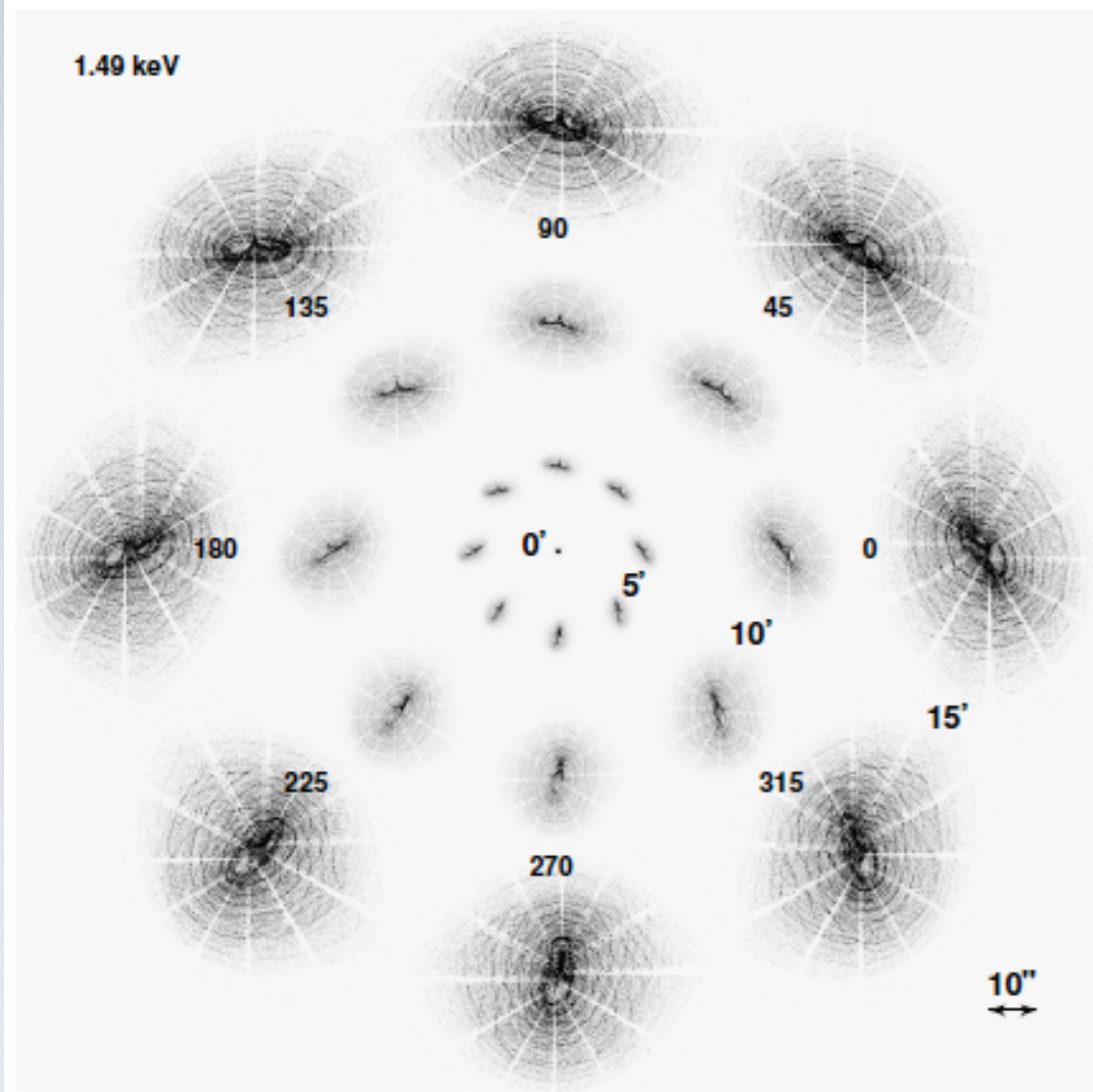


WOLTER TYPE I – SHORTENS
FOCAL LENGTH

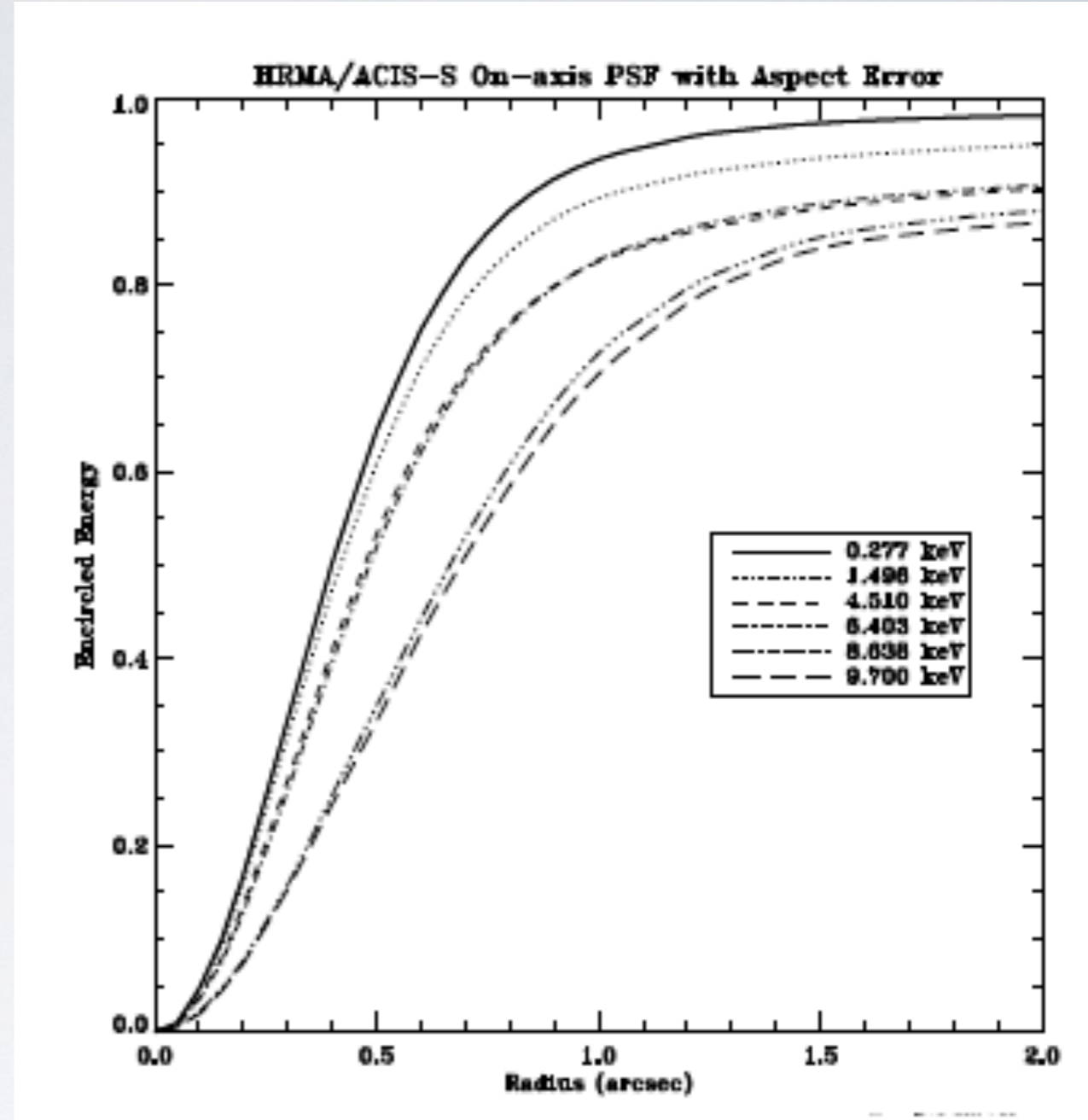
- Low Angle for Full Reflection
- Small Effective Area
 - Nested Mirrors
- Reflectivity exhibits atomic edge features of mirror materials
- Point Spread Function has strong dependence on angle & energy



*HRMA Effective Area vs. Energy
(with & without instruments)*

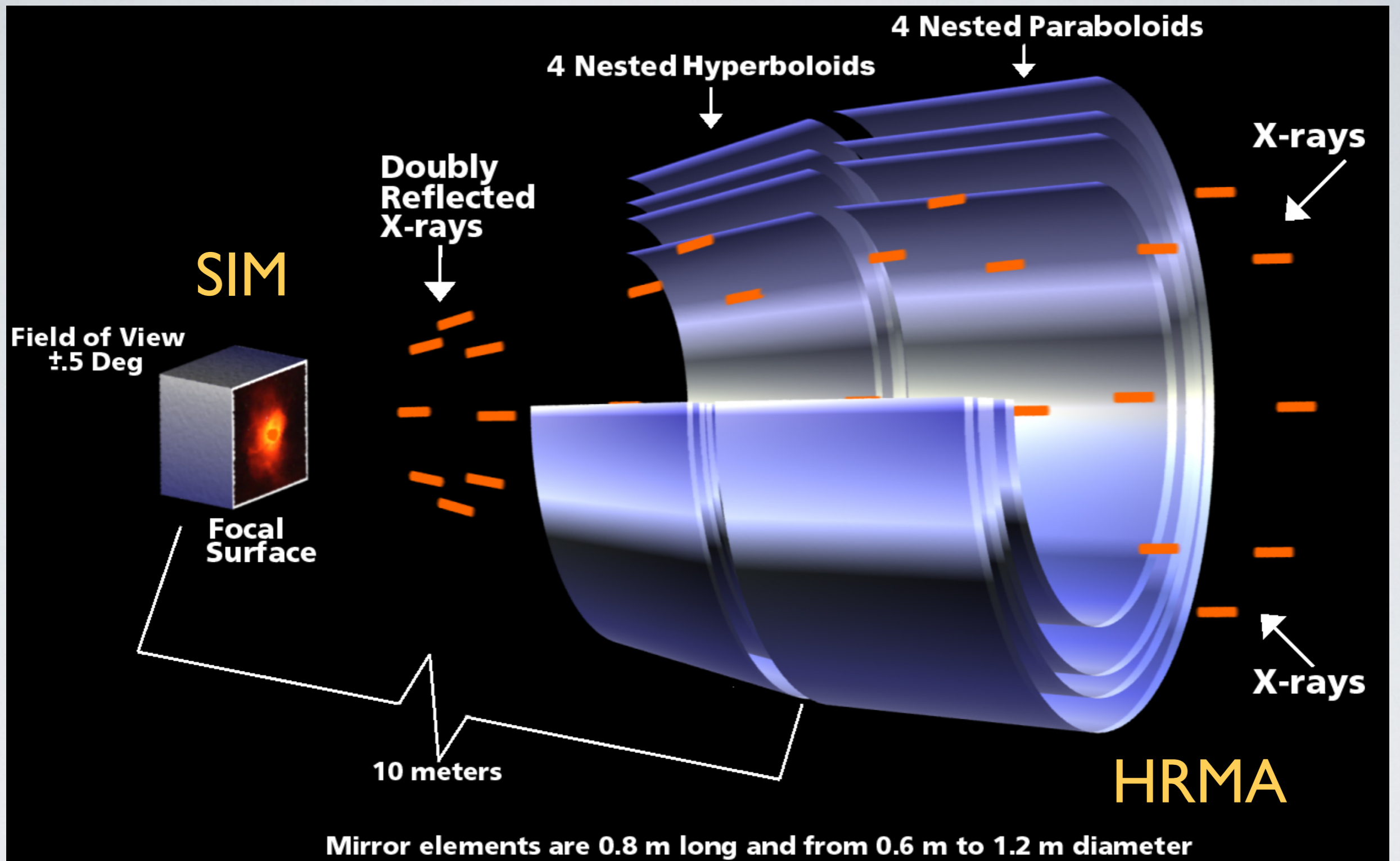


Chandra Point Spread Function



Chandra Encircled Energy Fraction

Chandra Spectrum Changes Depends Upon Detector Location and Width of Extraction Region!



MULTIPLE COORDINATE SYSTEMS
 (DETECTOR ← ASPECT → SKY)

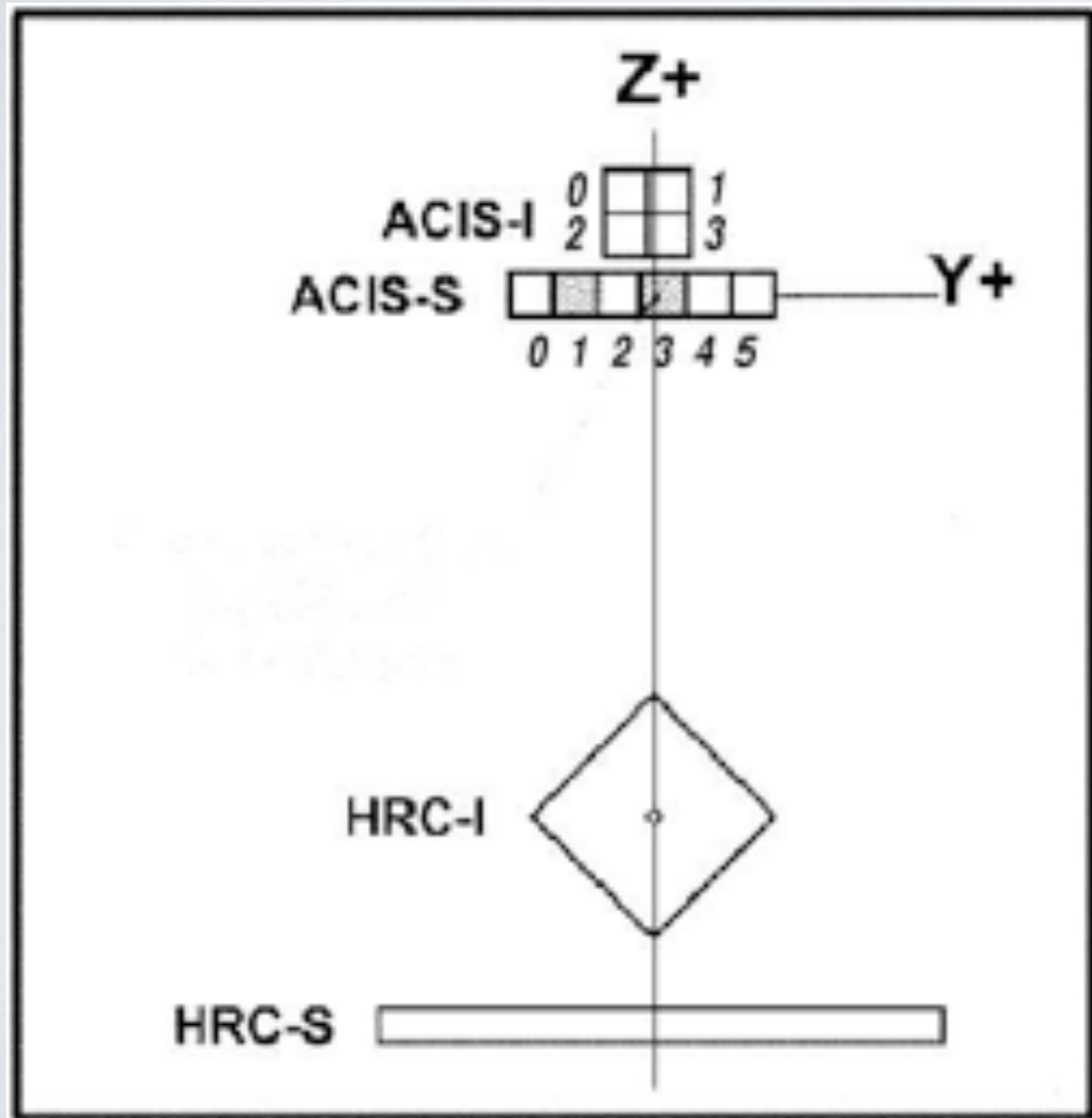
MIRROR GEOMETRY & STABILITY

- HRMA has been stable relative to SIM (i.e., stable optical axis and focus). SIM can be moved relative to optical axis (Y, Z) and focus (X);
- Star tracker relative to HRMA changes with time: aimpoint drift.
- We *reconstruct* our positions (detectors & spacecraft) very accurately, ($\approx 0.6''$) but *during* the observation, we can be off by up to $\approx 10''$.
- We intentionally dither our spacecraft $\pm 8''$, in two directions, so as to smooth out detector issues, and fill in the gaps between CCDs.

COORDINATE SYSTEMS

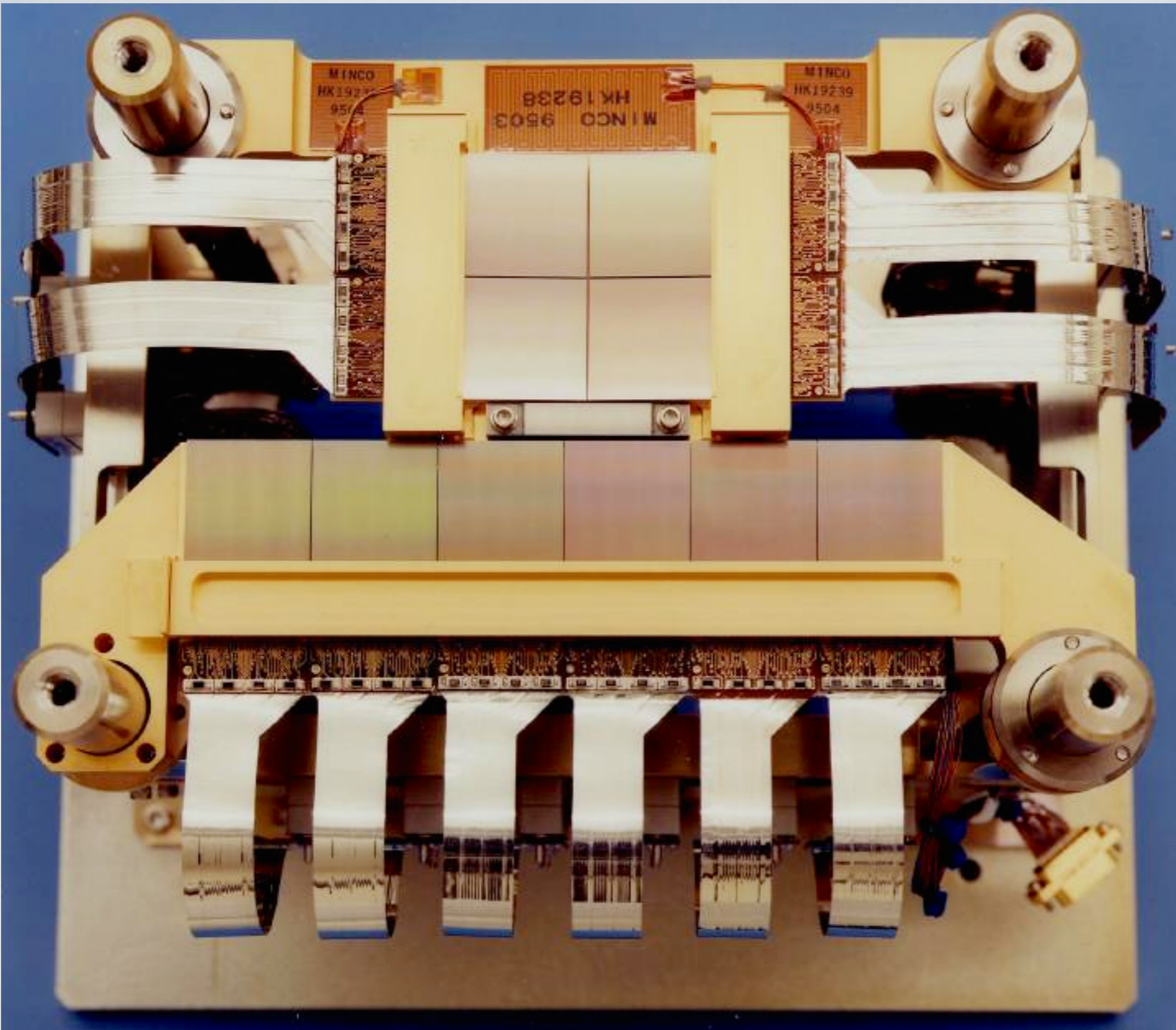
- Two Classes: Sky (e.g., X,Y) & Detectors (e.g., CHIPX, CHIPY)
- Special detector coordinates are used for gratings observations.
- The relationship between Sky, Mirrors, & Detectors is called the **aspect solution**.

SCIENTIFIC INSTRUMENT MODULE (SIM)



Advanced CCD for Imaging
Spectroscopy

High Resolution Camera

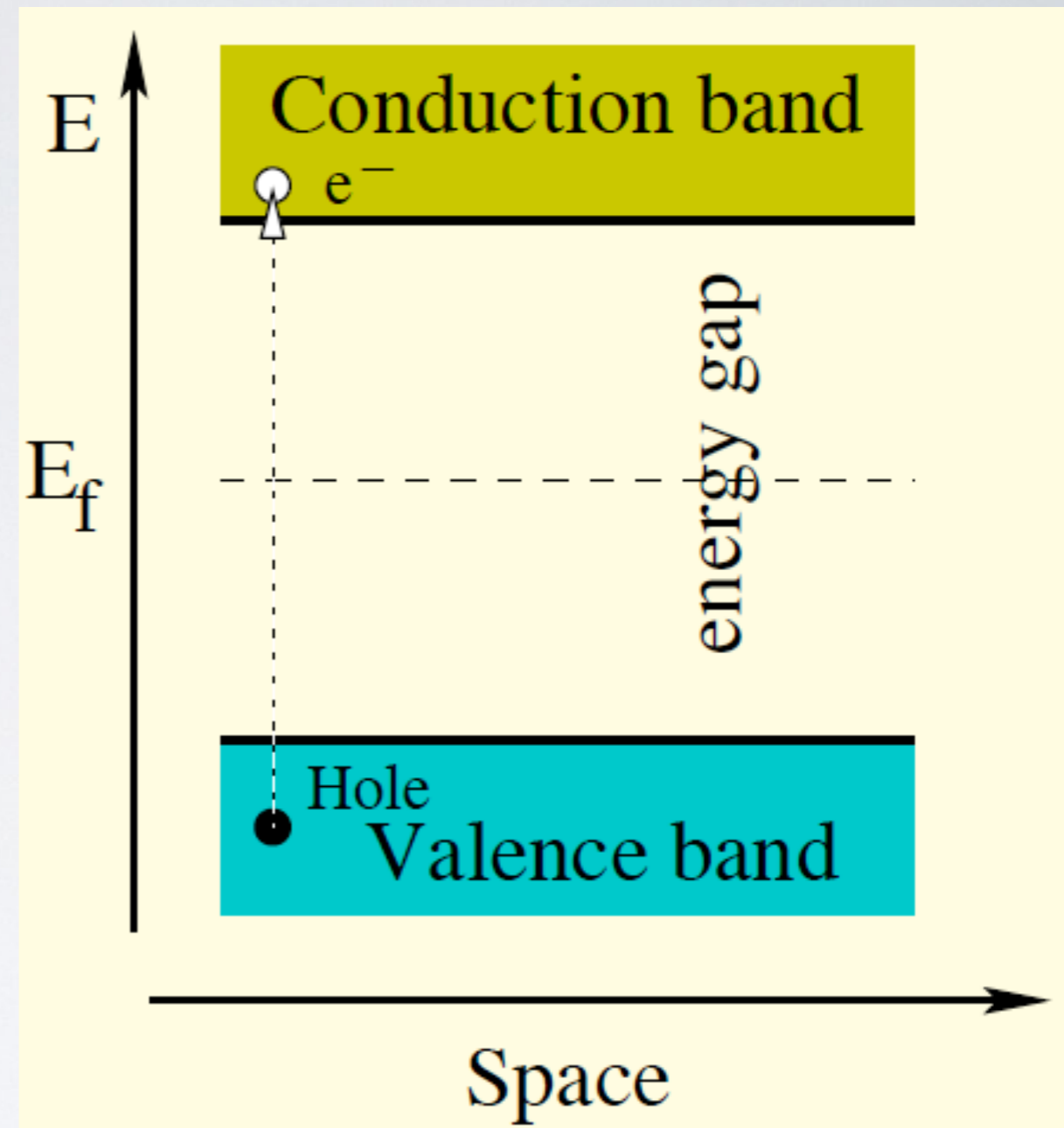


ACIS-I
16'x16'

ACIS-S
40'x8'

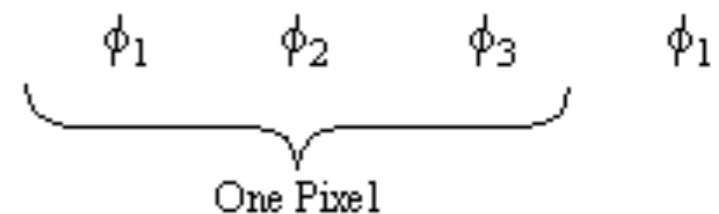
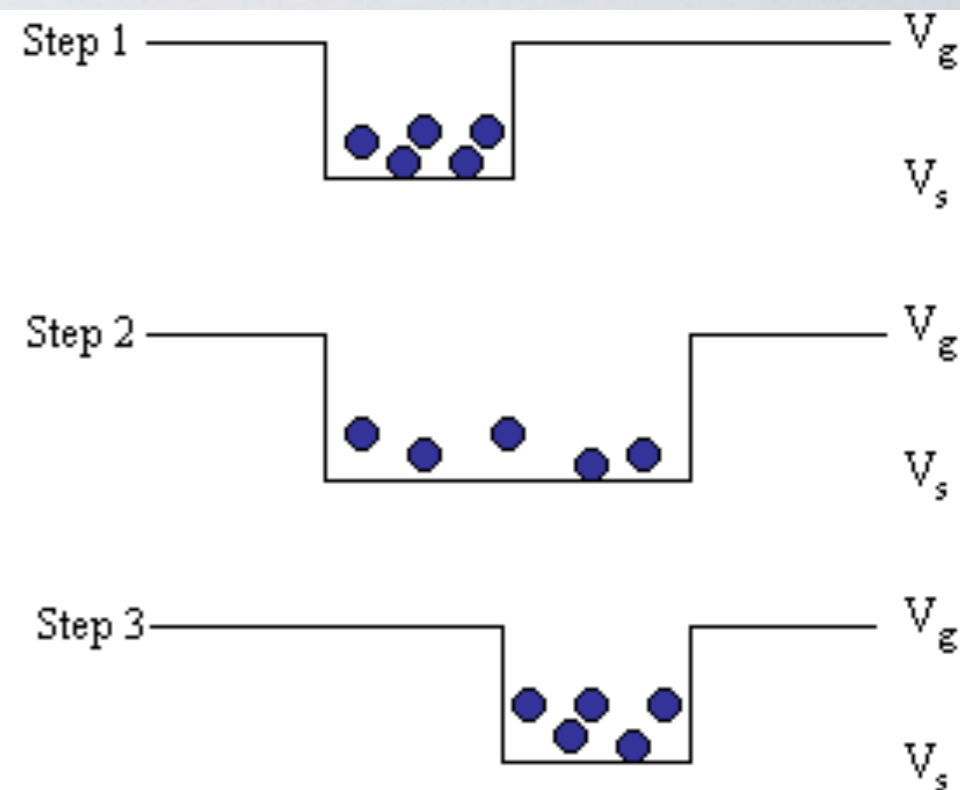
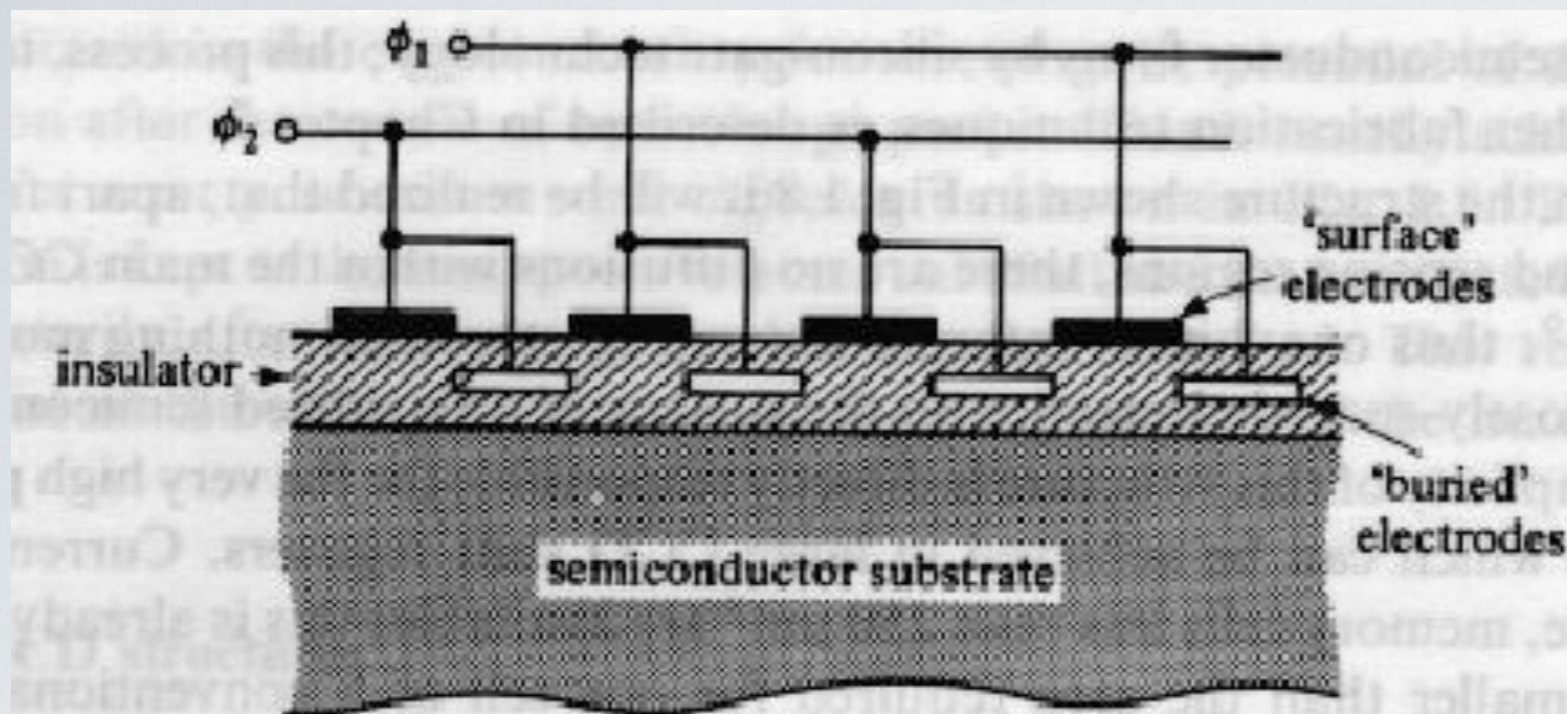
CCD DETECTORS

- Semi-conductor: absorption of a photon moves electrons across the energy gap between valence and conduction bands
- Number of electrons liberated go as: $N \sim h\nu/E_{\text{gap}}$
- $E_{\text{gap}} \sim 1\text{eV}$, so $N \sim 1000$ at 1keV .
Resolution: $\propto \sqrt{N} \propto \sqrt{E}$

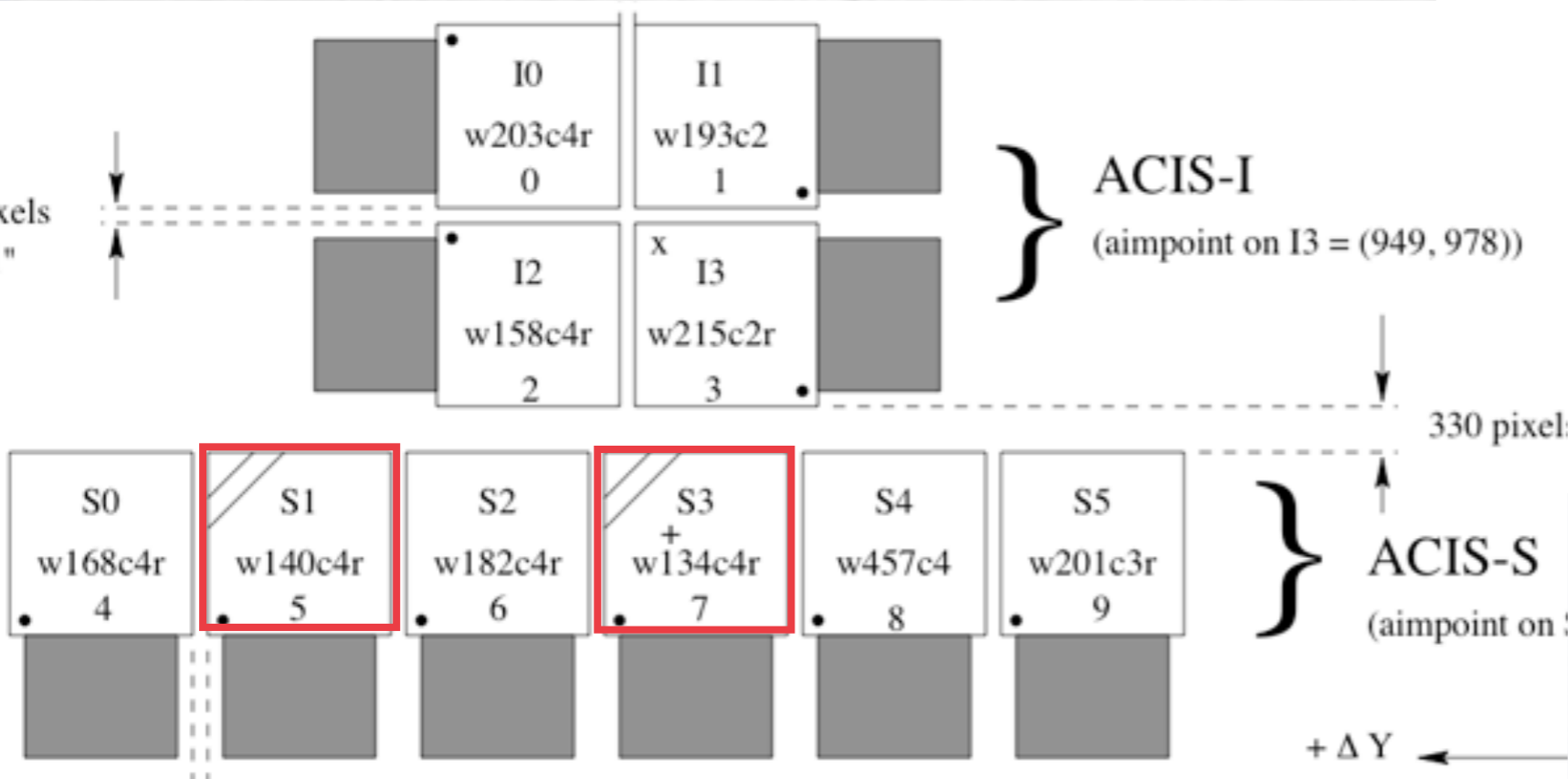


- CCD pixels are defined by physical structure and electrodes with applied alternating currents (“gate structure”)
- Applied voltage moves charge across pixels to readout

CCD DETECTORS



After every 3 steps, charge packet moves closer to the output by 1/3 of a pixel.



CCD_ID = 0–9; Chips I0–I4, S0–S5

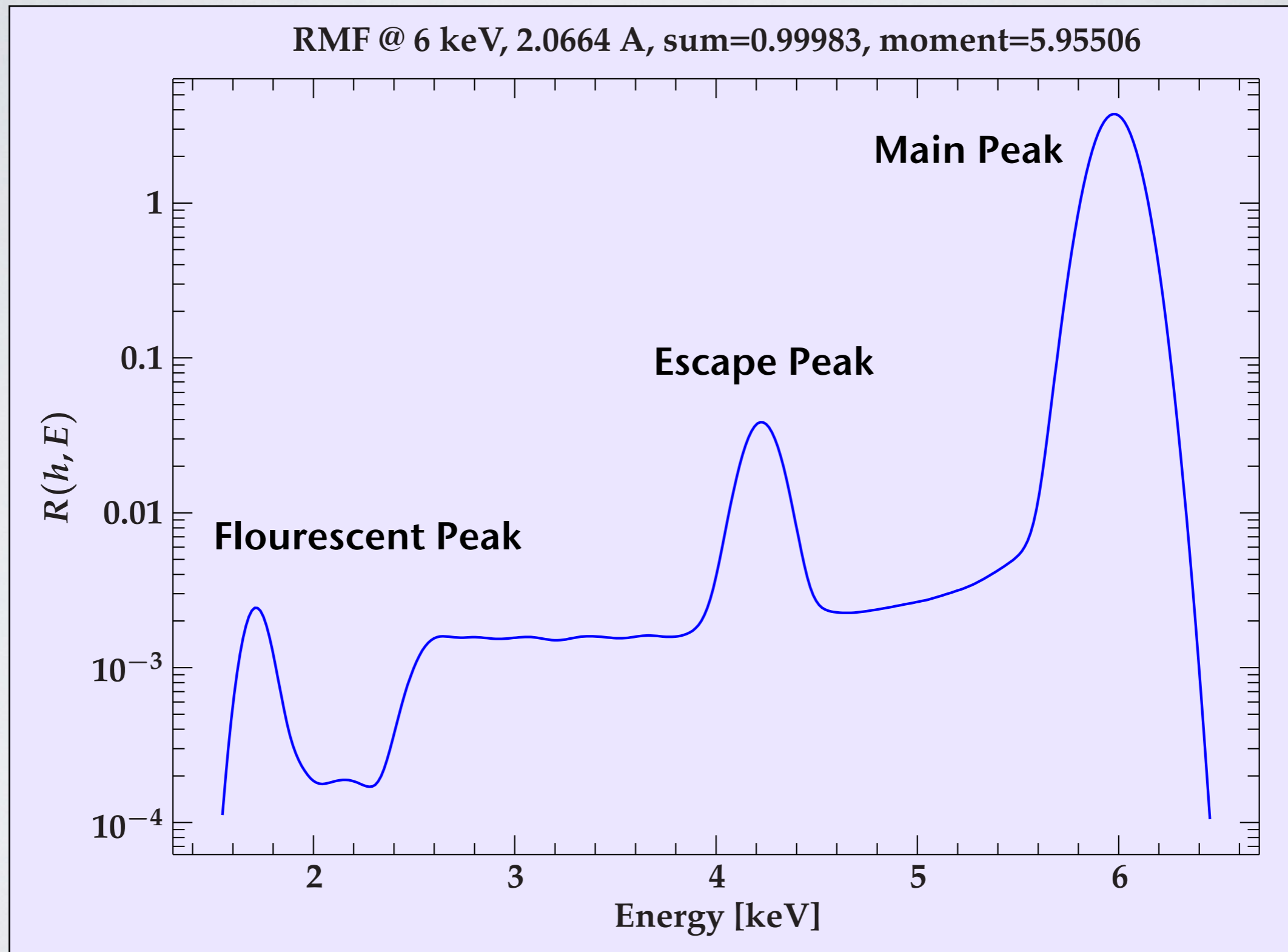
CCD DETECTORS

- Number of readouts could be: one (a row at a time), one per column, or one per pixel. Chandra = row at a time
- Charge is quickly transferred to “readout frame” (41.04 msec), where it is readout one row at a time (2.85 msec/row; 3.2 sec total) (Total: 3.24104 sec)
- Can decrease “Frame Time” by only reading out part of chip
- Can sacrifice one spatial dimension and read rows in “Continuous Clocking” (CC-) mode. 2.85 msec per row.
- In practice, CC-mode is (mostly) only used with gratings.

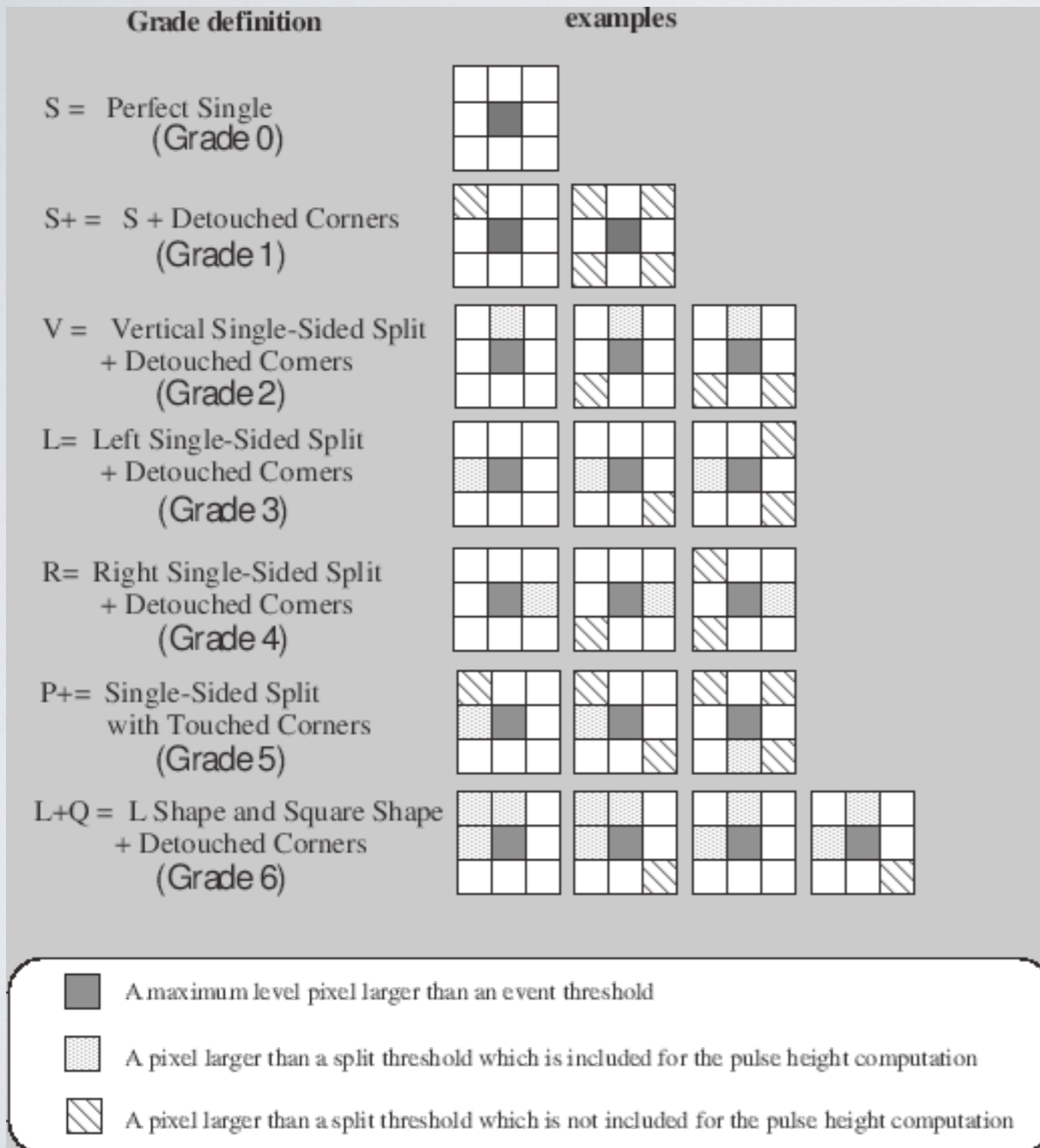
CCD DETECTORS

- Photons can interact with the detector material!
- If $E >$ Silicon K-edge energy (1.84 keV), K-shell ($n=1$) electron can be liberated and reabsorbed
- $L \Rightarrow K$ fluorescent line (1.74 keV)
- Free-bound & higher n -shell cascade transitions, with summed energy $= E - E_{\text{fluorescent}} = \text{Escape Peak}$
- Aside from resolution width, signatures can appear near fluorescent and “escape peak” energies!

CCD DETECTORS



CCD DETECTORS



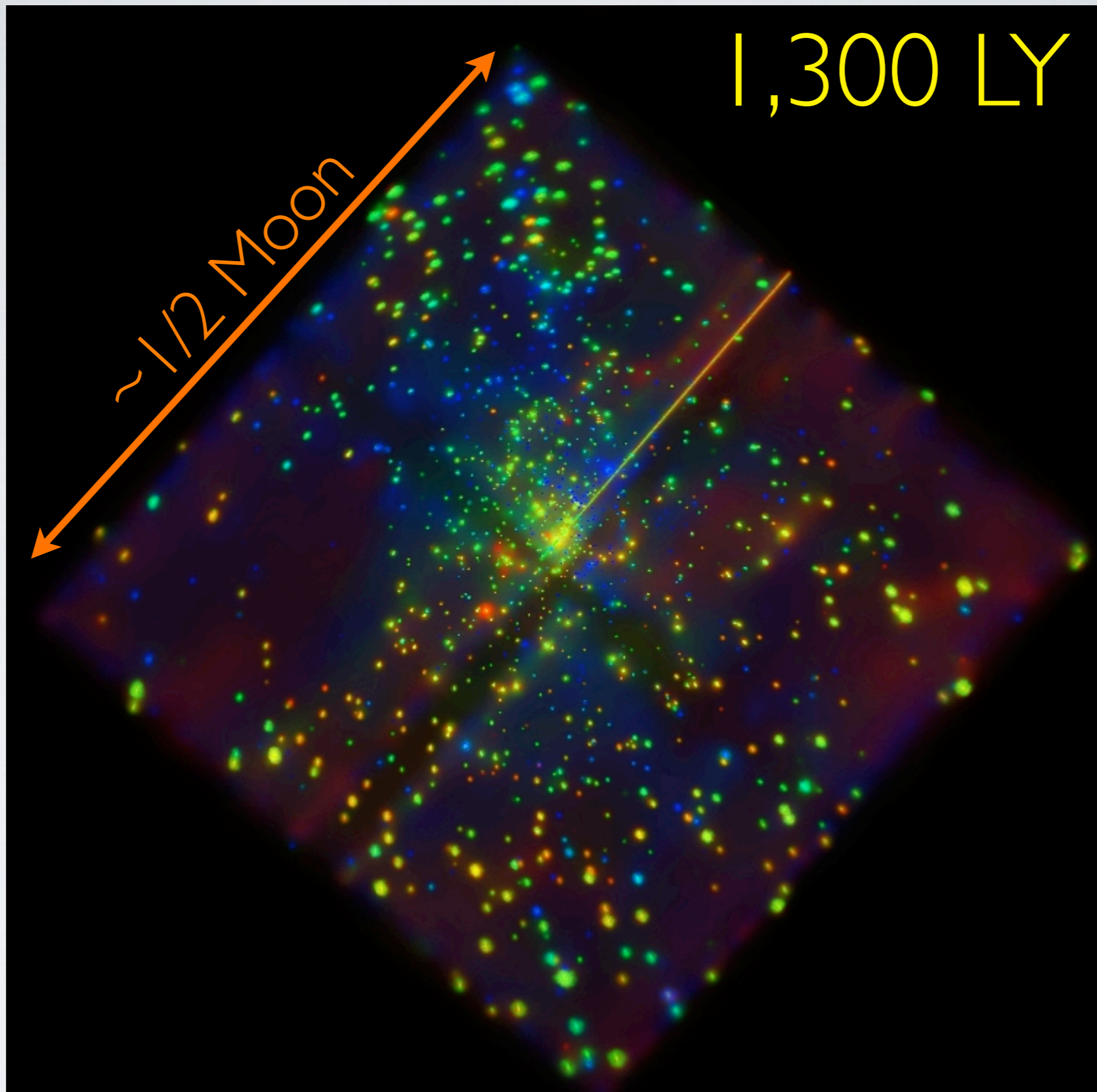
- Charge is typically deposited in more than one pixel.
- Chandra: 3x3 pixel regions \Rightarrow Grades.
- Bad grades more likely to be background (cosmic rays)

CCD DETECTORS

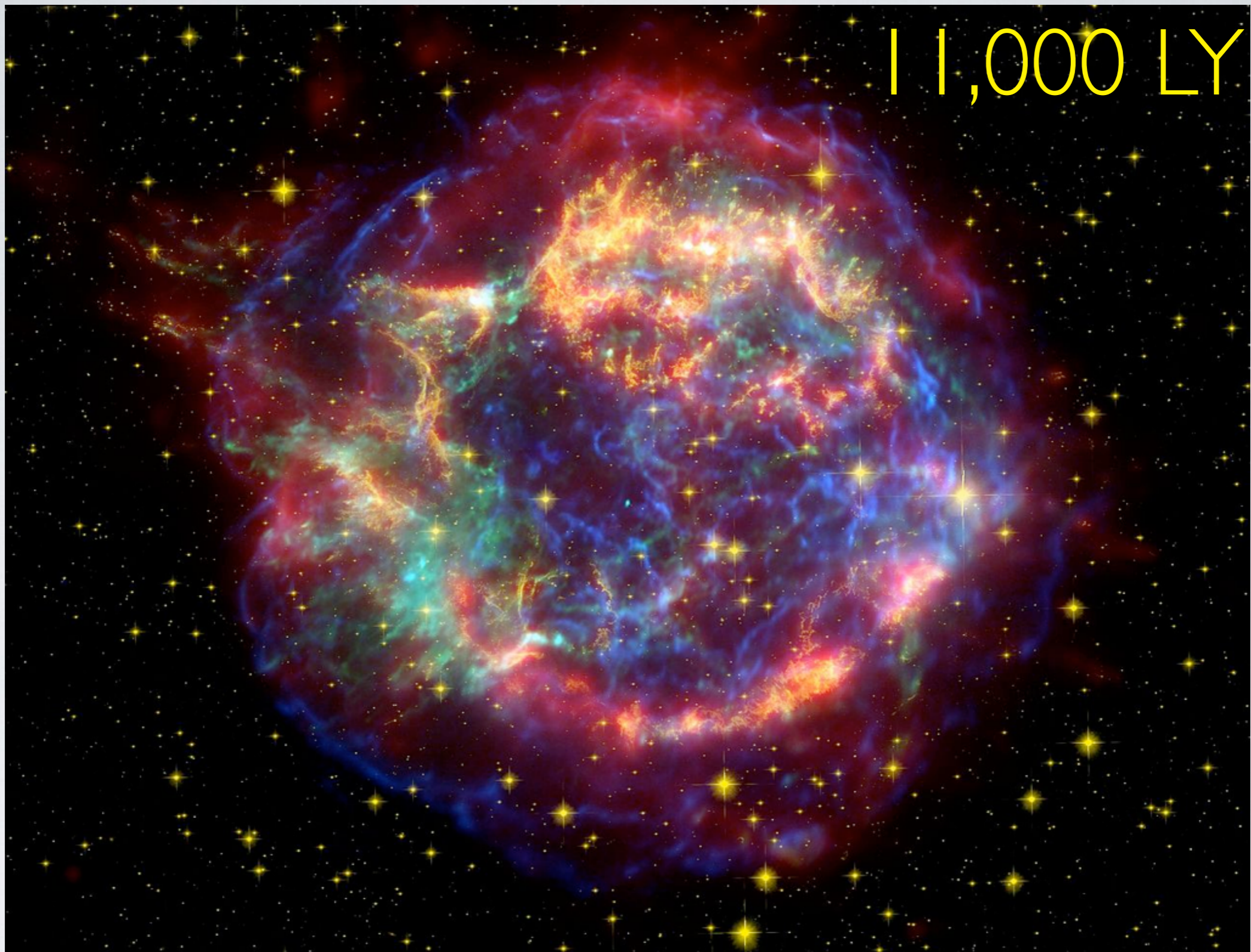
- One or more photons landing in the same region in the same frame = pileup.
- Piled photons are either read as a “bad grade”, or as an event with the summed energy!
- Pileup can be avoided by limiting the exposure:
 - Filters, i.e., inserting gratings
 - Reading only a fraction of the CCD imaging area to reduce readout times
 - “continuous readout” modes which sacrifice one spatial dimension of information

CCD DETECTORS

- CCD “gate structure” can be placed either facing the X-ray source (“front side illuminated” – more silicon), or away from the source (“backside illuminated” – less silicon).
- ACIS-S has two “backside illuminated” CCDs: S1, **S3**
- Backside illuminated is more sensitive to low energy photons, but has higher noise
- CCD pixels can be damaged, leading to “charge transfer inefficiency” (CTI). Charge is left behind as it is moved, leading to decreased spectral resolution in the detector
- Chandra has CTI damage, but we have tools to help correct it.

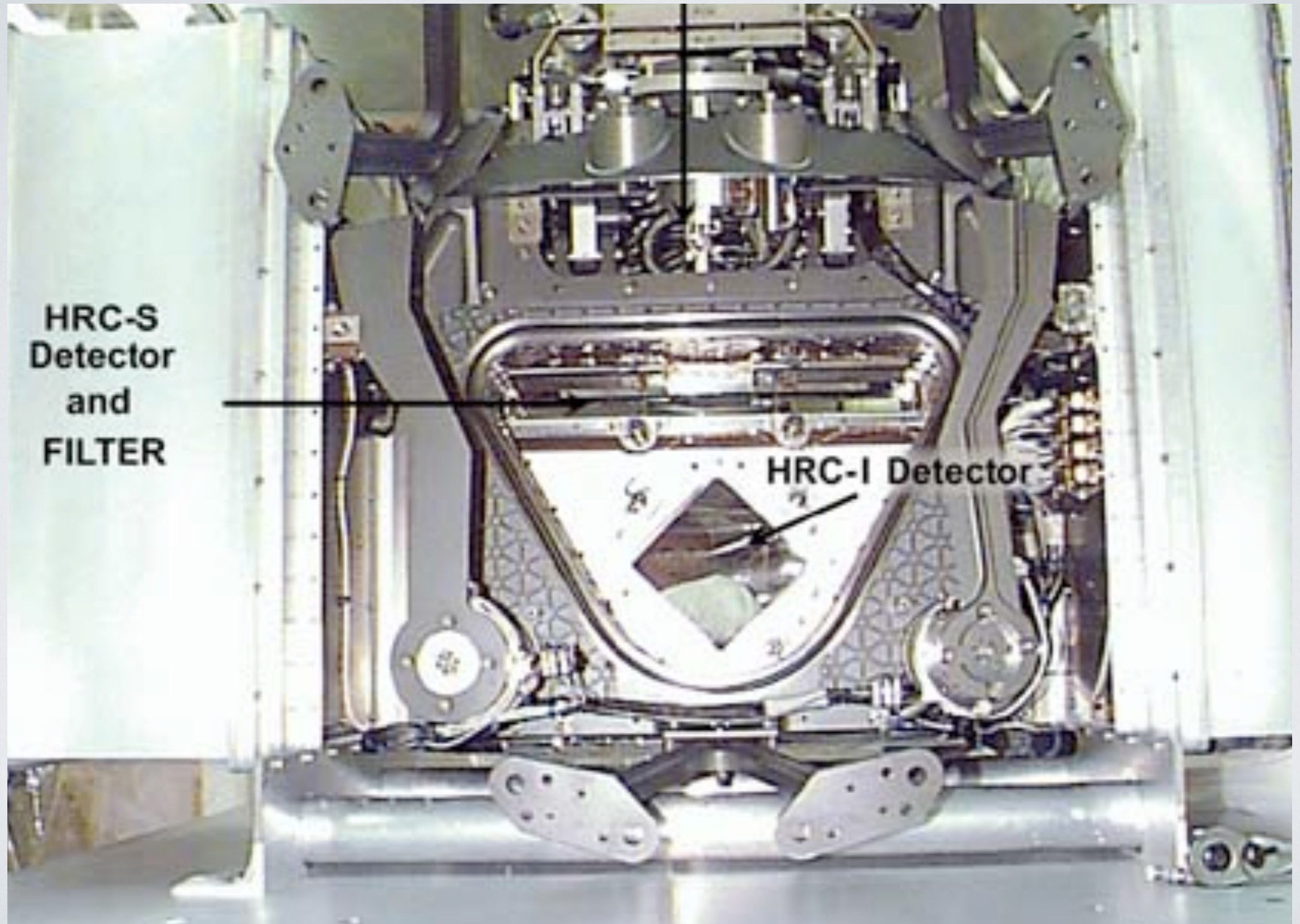


ORION STAR CLUSTER



CASSEOPEIA A SUPERNOVA
REMNANT

HRC-I: 30'x30'

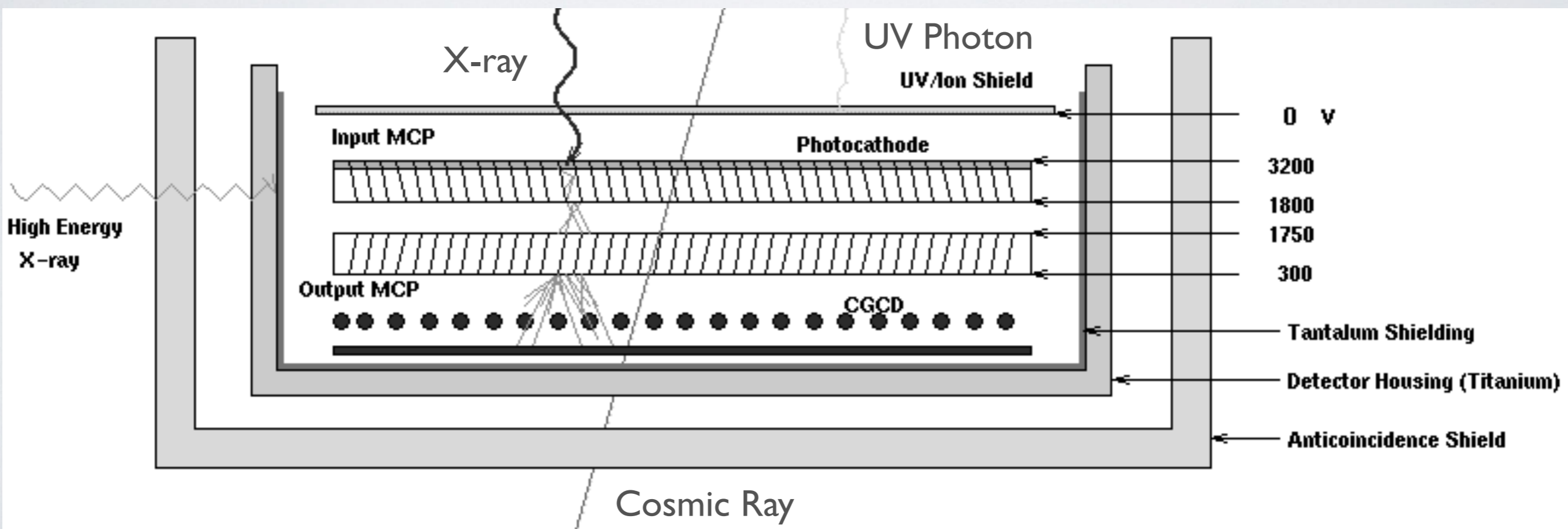


HRC-S
Detector
and
FILTER

HRC-I Detector

MICRO-CHANNEL PLATE

- X-rays interact with CsI coated, narrow & canted tubes, to produce photoelectrons that are then detected.
- Lower energy range than even ACIS-S, but essentially no energy resolution. (But also no pileup!)

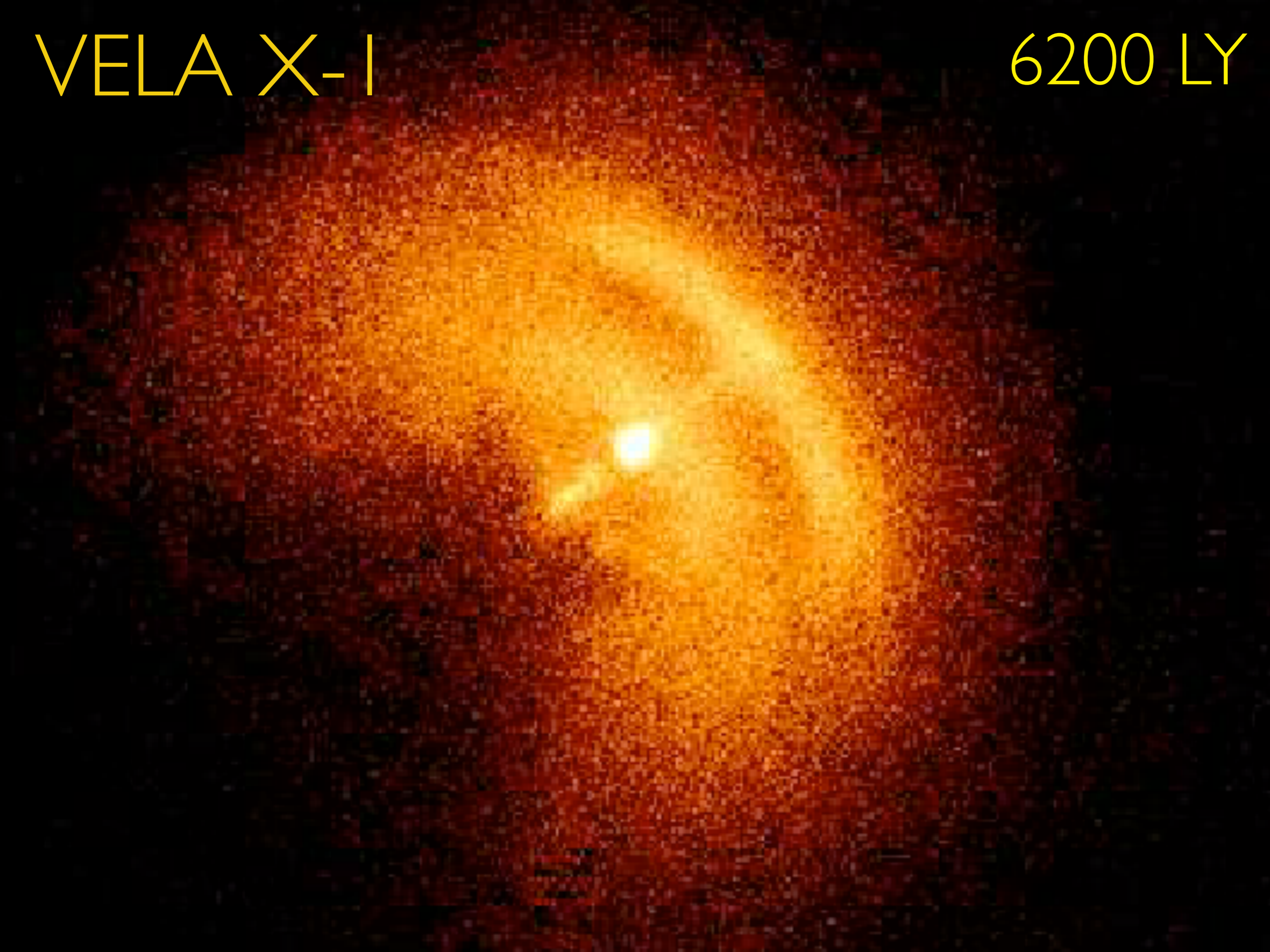


HRC

- Best Timing on Chandra – 16 microseconds in principle, but wiring error leads to time tag being for next event (\Rightarrow msec timing)
- Most likely coming from background event (rates \approx 180 cps).
- Recoverable if no events are lost. HRC-S timing mode uses only 6'x30' detector region to limit background rate
- HRC is typically used in very crowded fields where best position is necessary, and/or when timing information is necessary
- E.g., localize an accreting, X-ray msec pulsar

VELA X-1

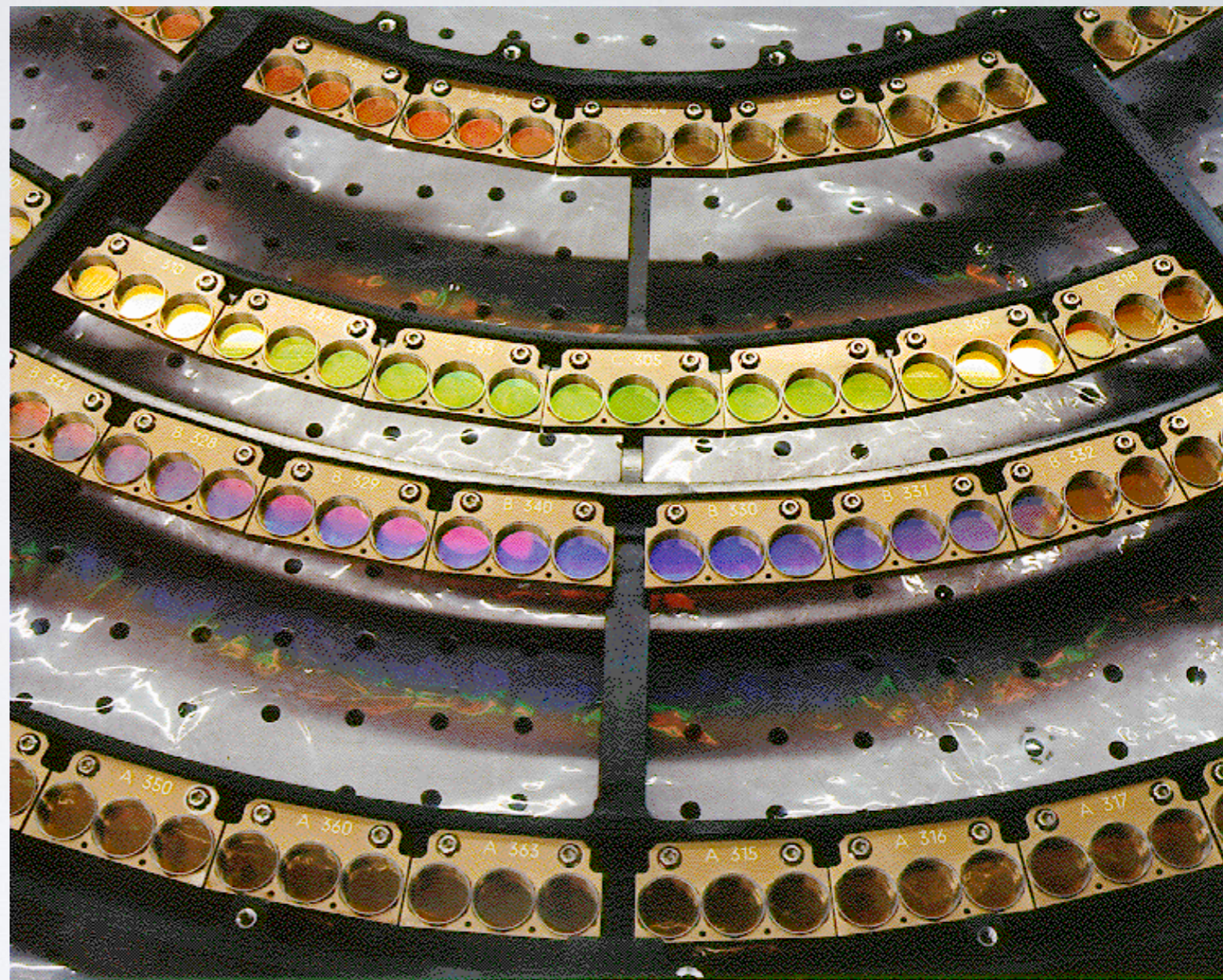
6200 LY



TRANSMISSION GRATINGS

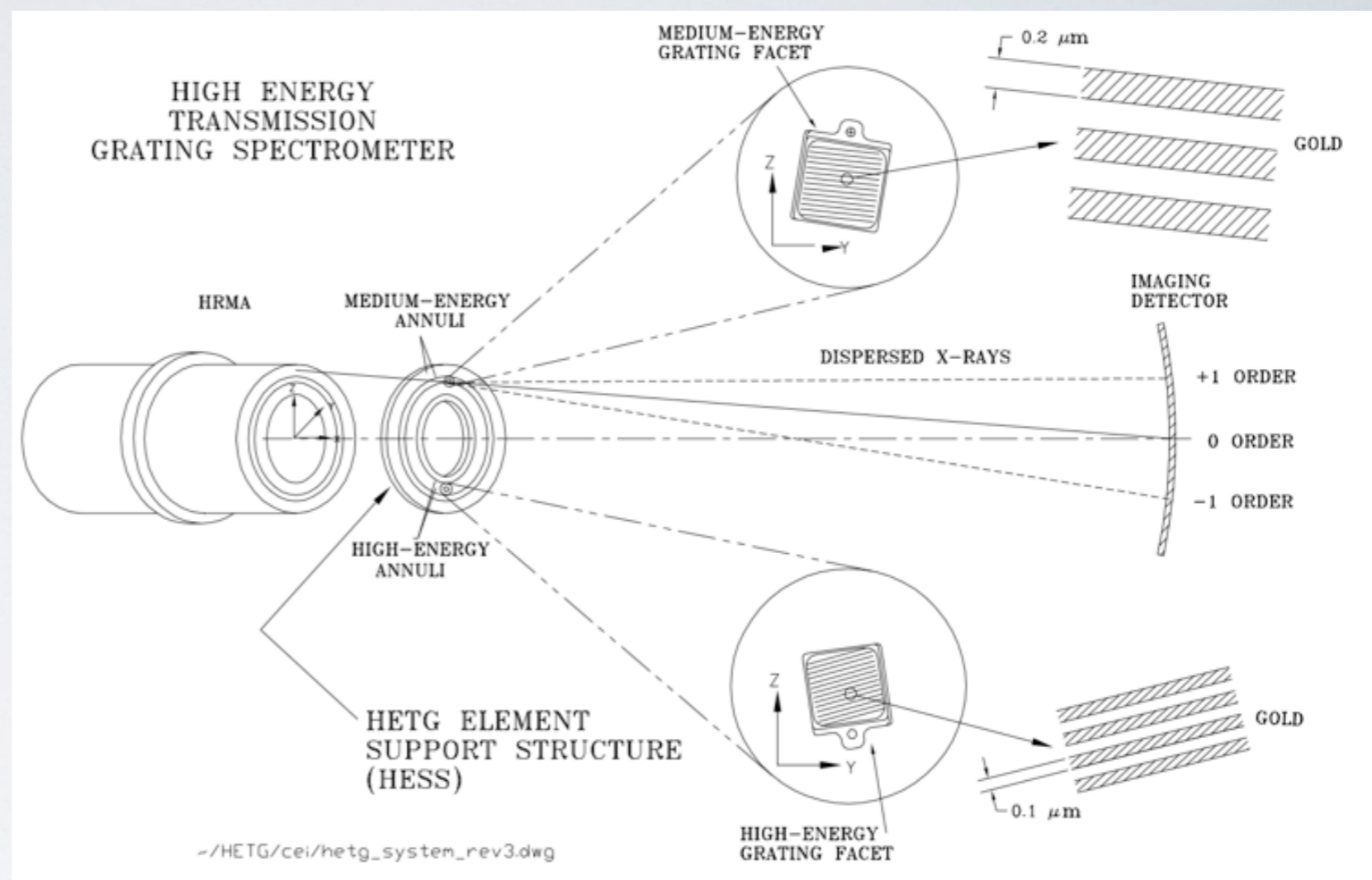
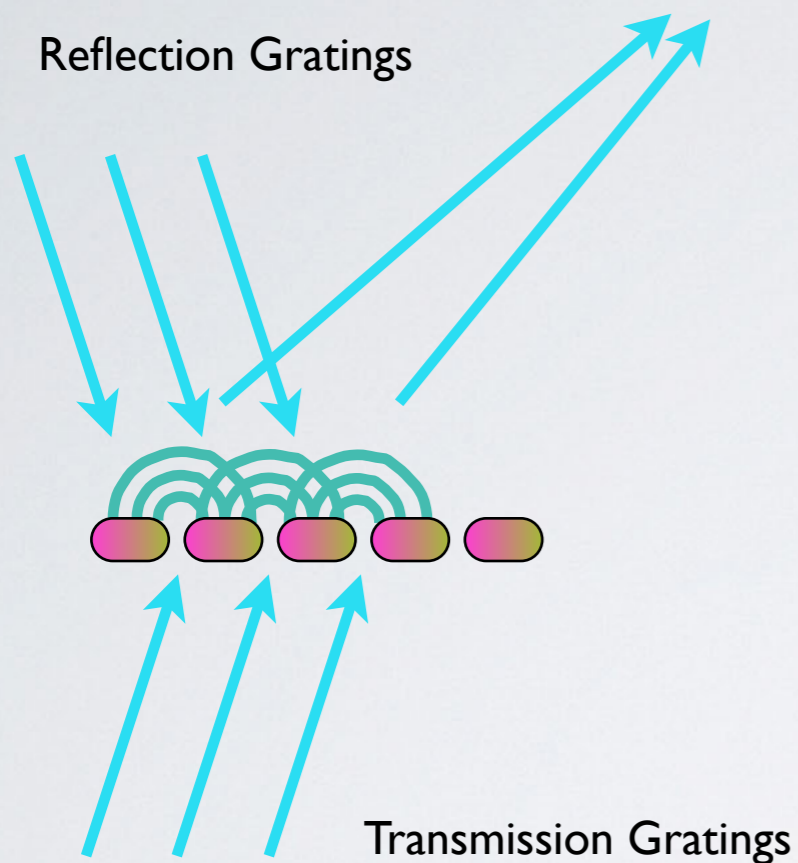


High Energy
Transmission Gratings



Low Energy
Transmission Gratings

GRATINGS

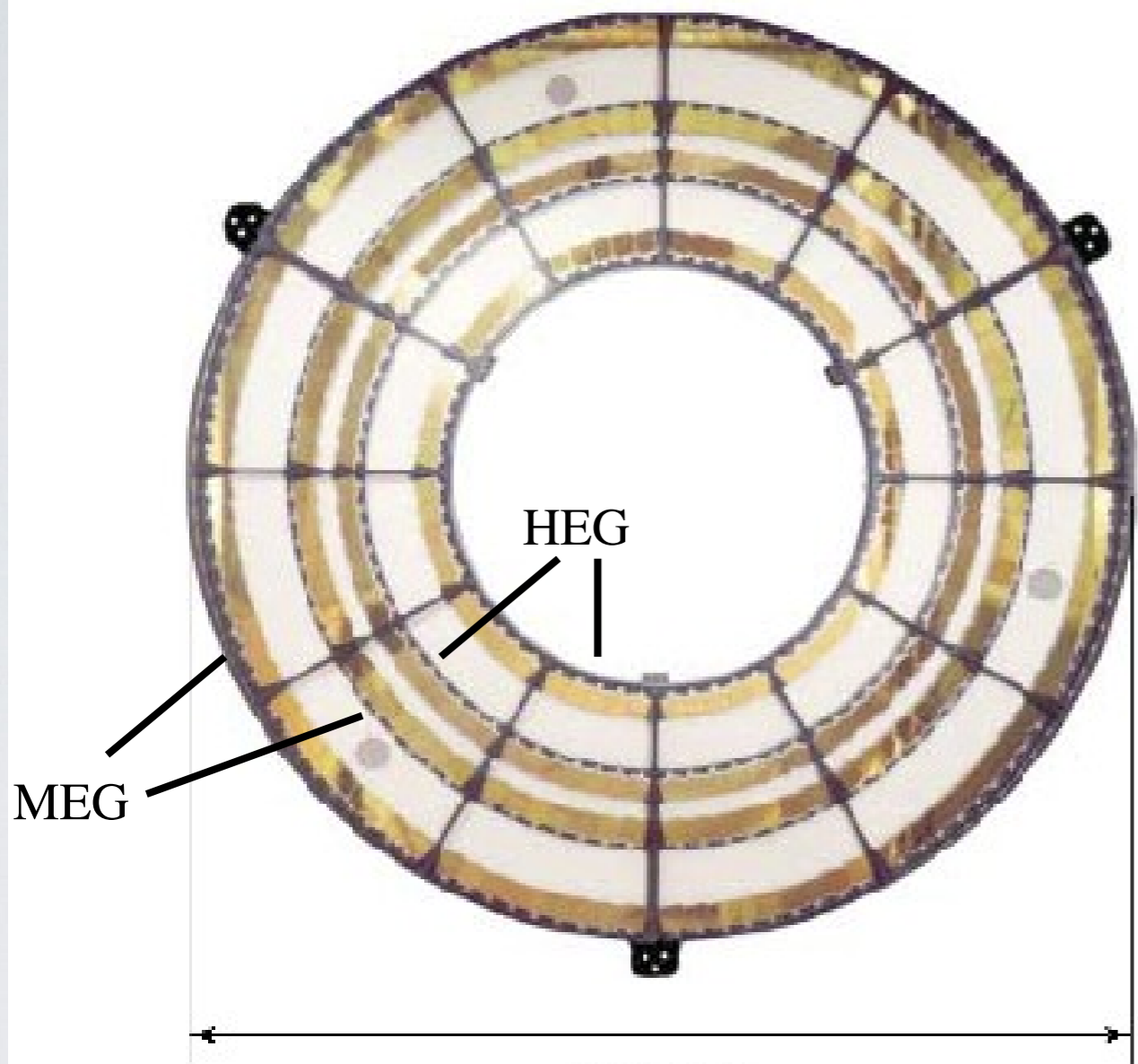


Gratings Equation:

$$m\lambda = m \frac{hc}{E} = p \sin \approx p\beta$$

CHANDRA-HETG

HETGS instrument.



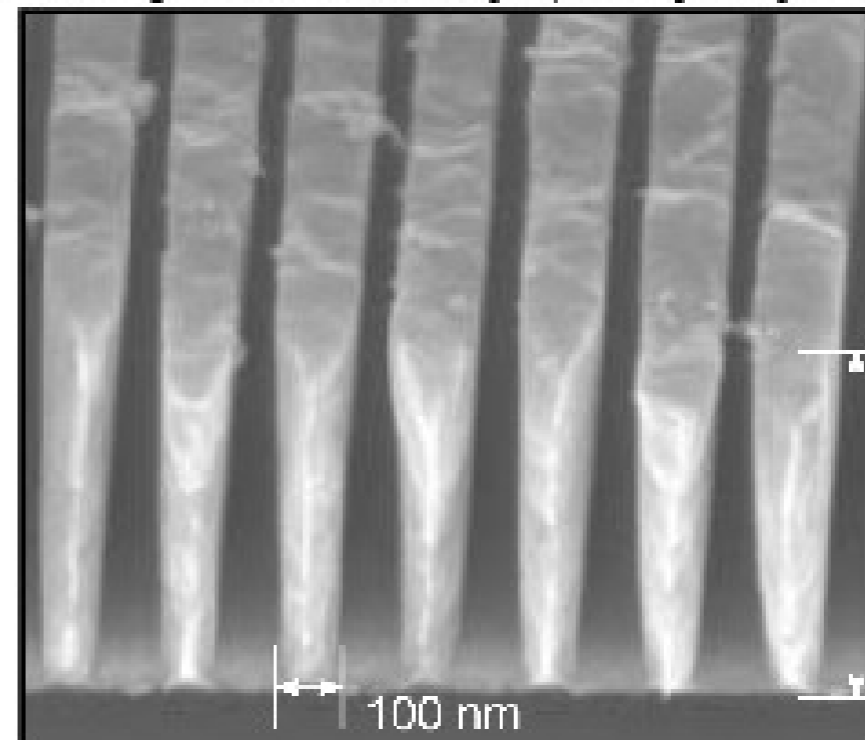
1.1 meter

1/1000 of a Human Hair

Invar grating frame.



Scanning electron micrograph of gold grating.



550 nm

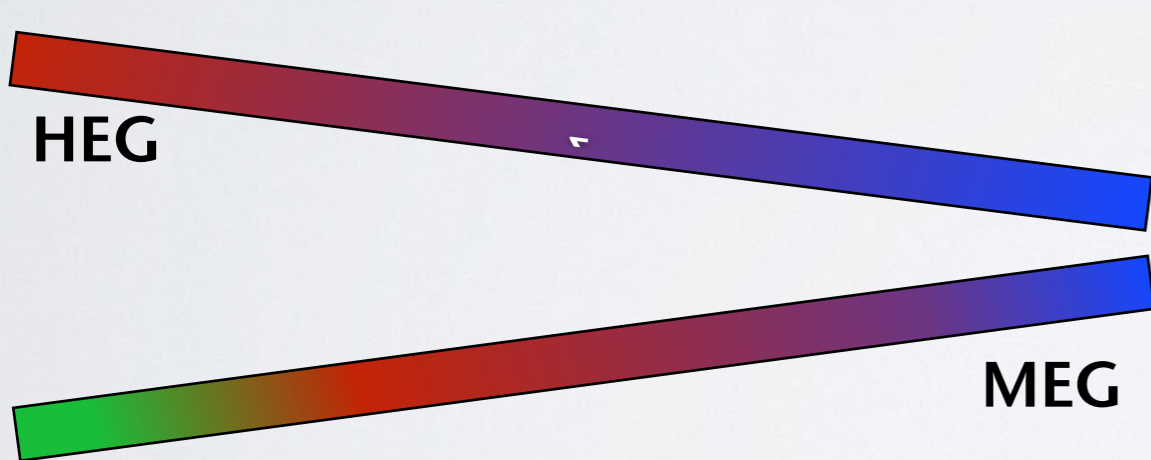
100 nm

$$m\lambda = m \frac{hc}{E} = p \sin \approx p\beta$$

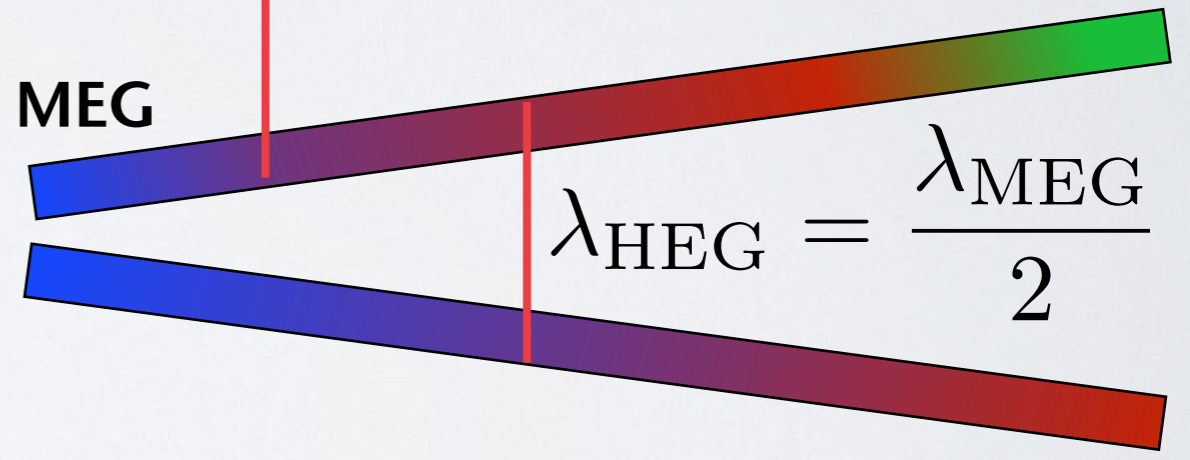
LETG



HETG



MEG



$$\lambda, \frac{\lambda}{2}, \frac{\lambda}{3}, \dots$$

$$E, 2E, 3E, \dots$$

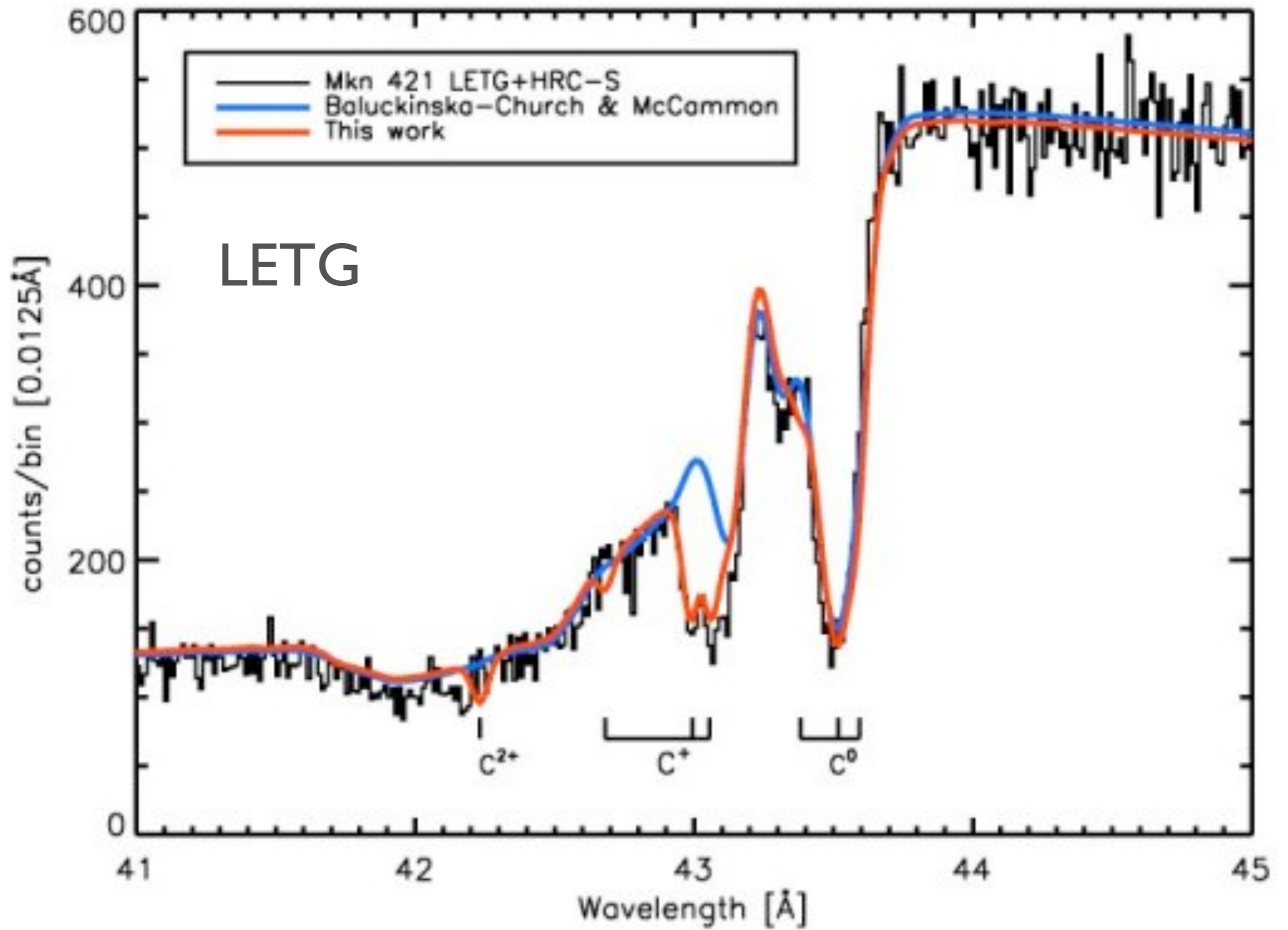
$$\lambda_{\text{HEG}} = \frac{\lambda_{\text{MEG}}}{2}$$

HEG

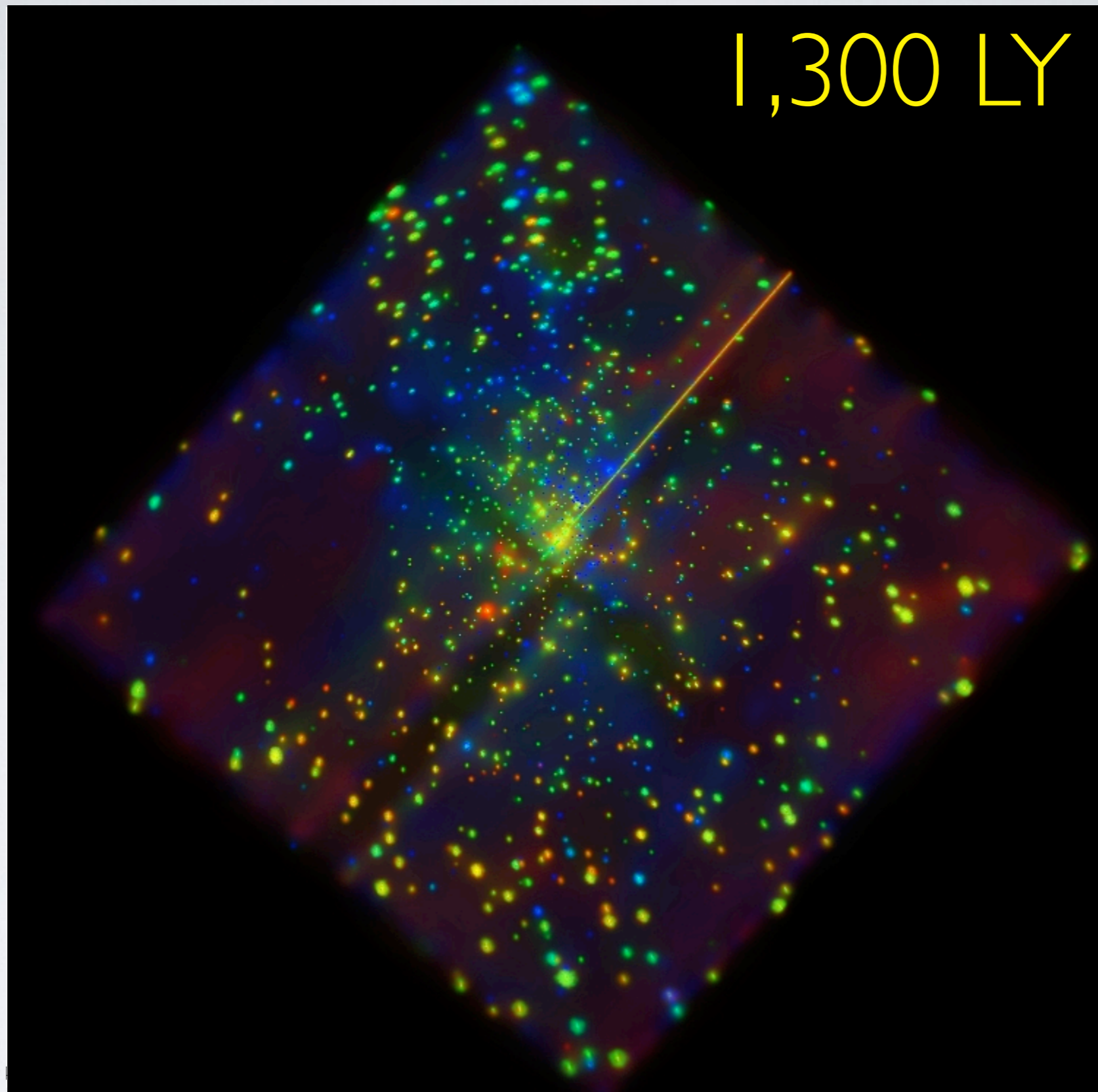
Greater Distance = Higher Resolution
 Resolution Limited by CCDs & Gratings Accuracy

TRANSMISSION GRATINGS

- HETG & LETG can be used with any combination of ACIS-S/-I and HRC-S/-I (-S array geometries optimized for specific gratings)
 - Almost all are: HETG/ACIS-S, LETG/HRC-S, or LETG/ACIS-S
- ACIS-S has CCD energy resolution, HRC-S no energy resolution
 - Can separate orders on ACIS-S, cannot with HRC-S
- Highest resolution is ACIS-S/HETG (with HEG), longest wavelength/lowest energy is HRC-S/LETG
- Gratings lowers the effective area; however, brightest objects observed by Chandra are all CC-mode, HETG/ACIS-S.



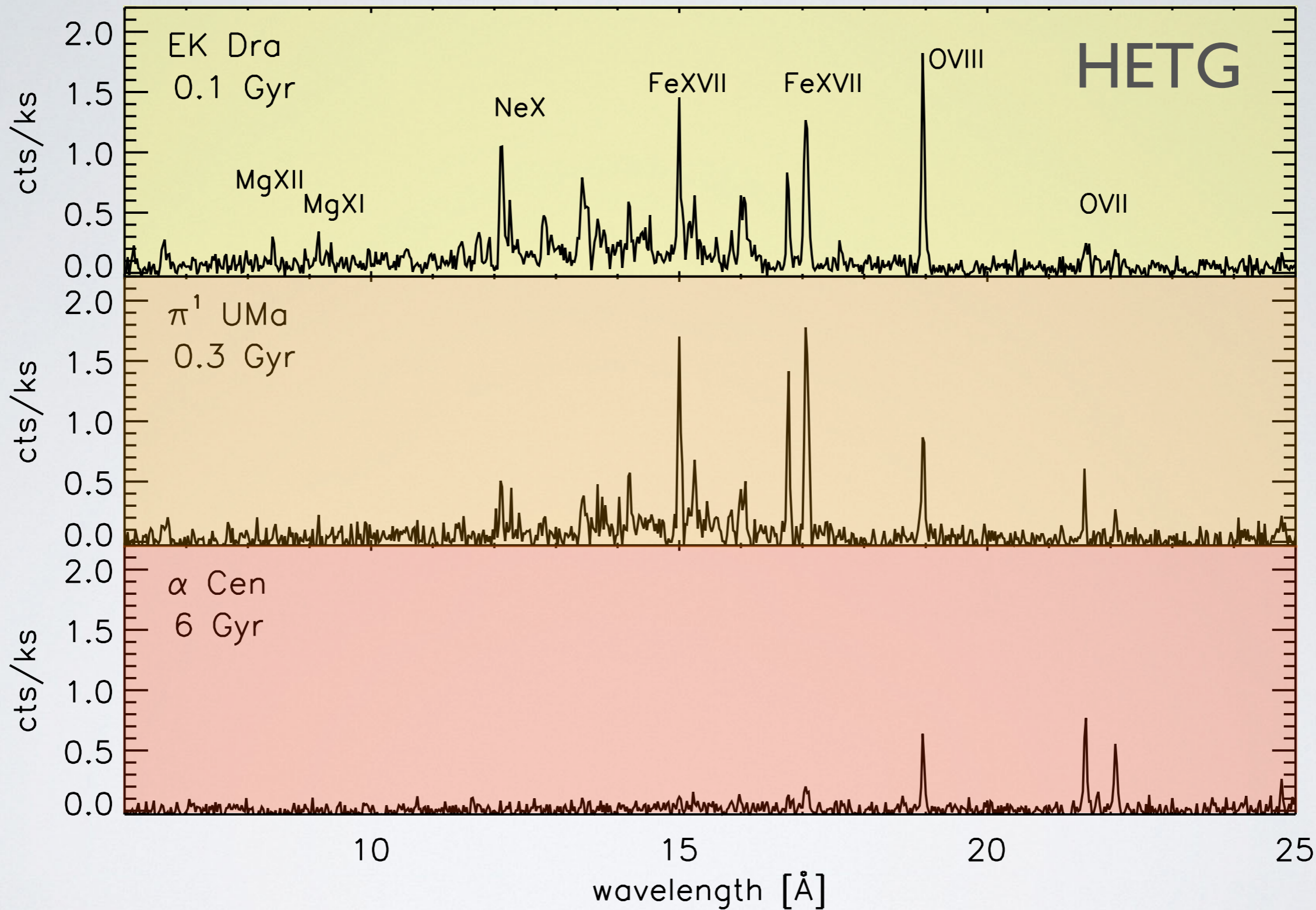
MKN 421, ISM CARBON EDGE



ORION STAR CLUSTER

← Energy

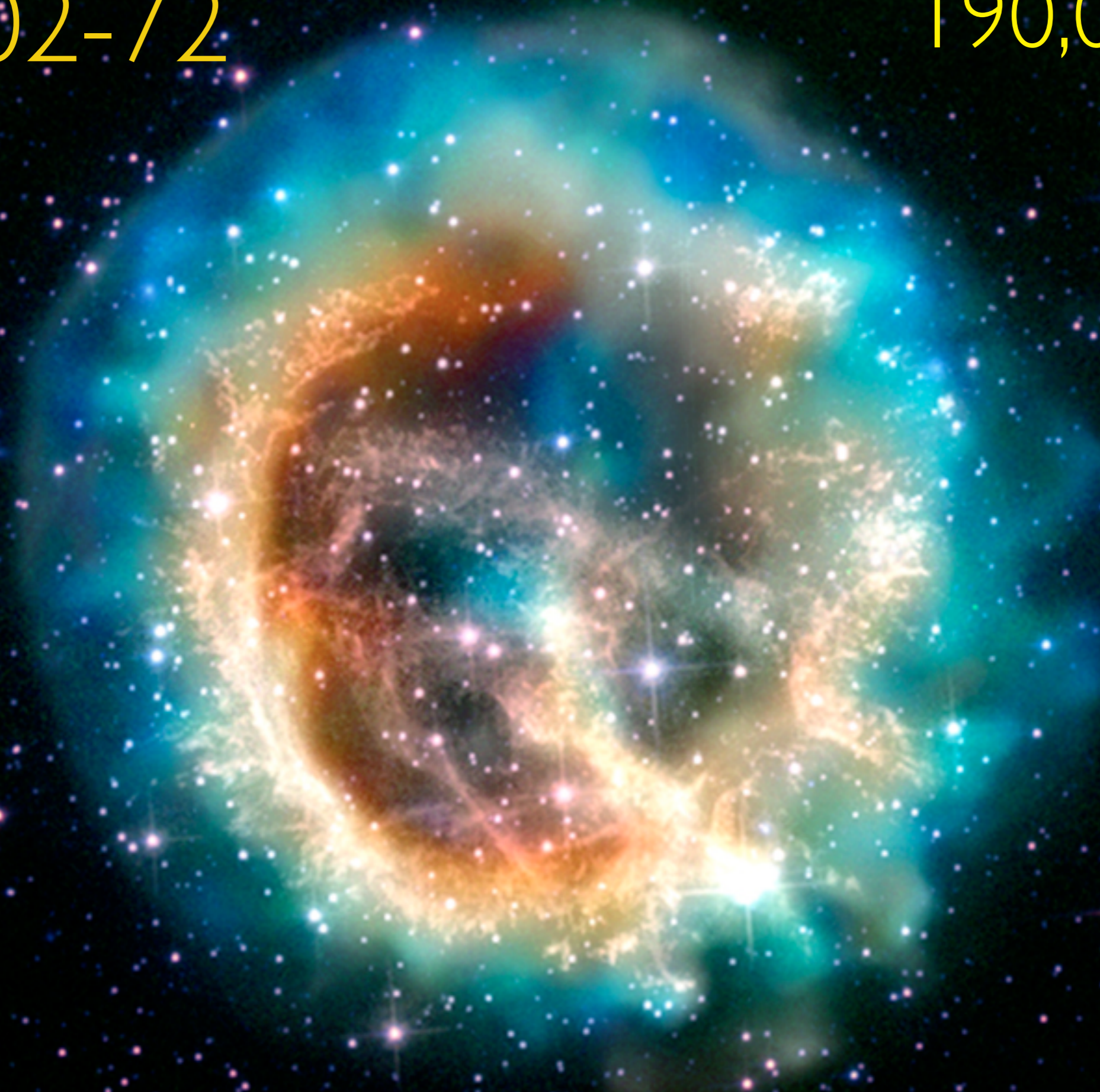
Age ↓



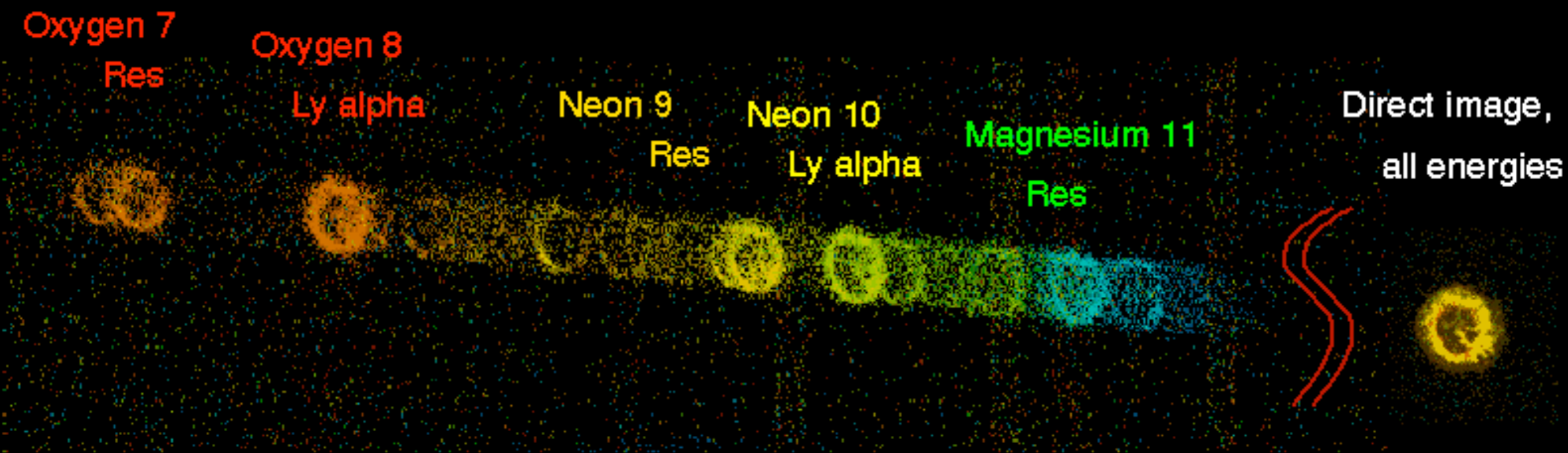
A “HISTORY” OF OUR SUN

E0102-72

190,000 LY



E0102-72



E0102-72

MIT/HETG

YEARLY OBSERVATION CYCLE:

- Guest Observer (GO) Program: ~17 Msec
 - Propose ~March, Reviewed ~June, starts ~October–January
- Guaranteed Time Observations (GTO): ~2 Msec
 - Instrument teams, without duplicating GO
- Calibration: ~1 Msec
- Director's Discretionary Time (DDT): ~700 ksec
 - (Typically) Unanticipated events & unique opportunities.

OBSERVATIONAL CONSTRAINTS:

- How long per orbit is the object visible?
- Where is it relative to the sun (& moon)?
- Thermal constraints (alternate between “hot” & “cold”):
 - We want to keep the CCDs operating at -120°C (CTI increases with temperature)
 - We want to keep fuel lines from freezing.
 - Schedule 1 week at a time; thermal balance affects whole week
- Observations can run from 1 ksec to 170 ksec. The latter is more difficult in these later years of the mission.

TYPES OF OBSERVATIONS WE DO

- ACIS-I Imaging: ~40%, ACIS-S Imaging: ~40%,
HETG/ACIS-S Spectra: ~18%, (HRC-S or ACIS-S)/LETG:~2%
- Science Topics:
 - Stars ~13%
 - Black Hole/Neutron Star Binaries ~9%
 - Galaxies, X-ray Populations, Surveys, & Diffuse Emission ~7%
 - Active Galactic Nuclei & Quasars ~23%
 - Clusters of Galaxies ~32%

LOTS MORE LEFT TO DO,
SO COME JOIN US!